

Advanced Application 19

**Construction Stage Analysis for
FSM (Full Staging Method)
using general functions**

Civil

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Outline

FSM (Full Staging construction Method) is a very basic method in constructing post-tensioned concrete bridges. Dead weight of concrete, formwork and falsework are fully shored over the full spans of a bridge until the concrete gains a certain level of strength.

FSM can be economical if the horizontal alignment of a bridge is curved or the width of the bridge deck widens, provided that the height of the piers are not too high.

In the case of a bridge with long spans, the use of continuous tendons can be limited, thereby requiring construction joints. Each segment may be constructed sequentially span by span. Structural analysis is carried out on the basis of construction stages defined by the construction joints. Although a bridge is supported by shoring, FSM is generally analyzed with the assumption that effect of support is negated by the effect of prestressing.

When FSM is applied to a bridge with continuous spans, the first stage is a simple span, and it becomes continuous with the progress of the construction stages. In comparison with an analysis that does not consider construction stages, the construction stage analysis results in lower negative support moments higher positive span moments. As such, a bridge constructed by FSM needs to be analyzed with construction stages reflecting both the change in structure, element load and boundary conditions as well as time-dependent material properties, including creep, shrinkage and modulus of elasticity.

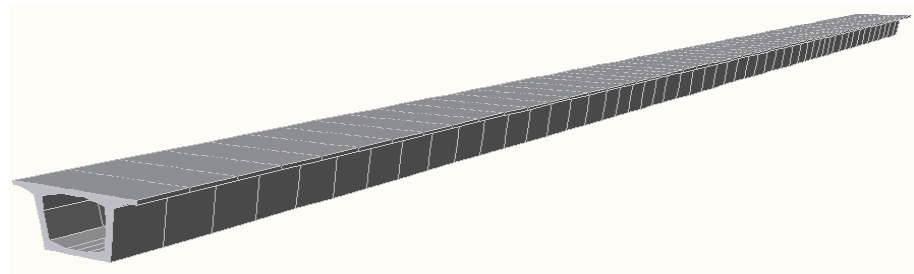


Figure 1. Bridge to be analyzed

Bridge profile and general section

This example has been simplified from an actual project for the purpose of illustrating construction stage analysis using FSM.

The bridge profile is defined as follows:

Structure type: 3 continuous span PSC Box girder bridge (F.S.M)
 Spans: $L = 40.0 + 45.0 + 40.0 = 125.0\text{m}$
 Bridge width: 8.5m
 Skew angle: 90°

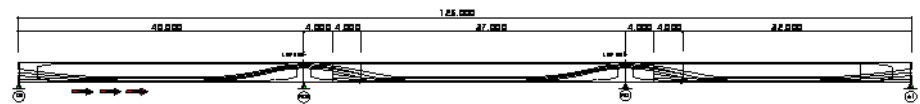


Figure 2. Longitudinal Section

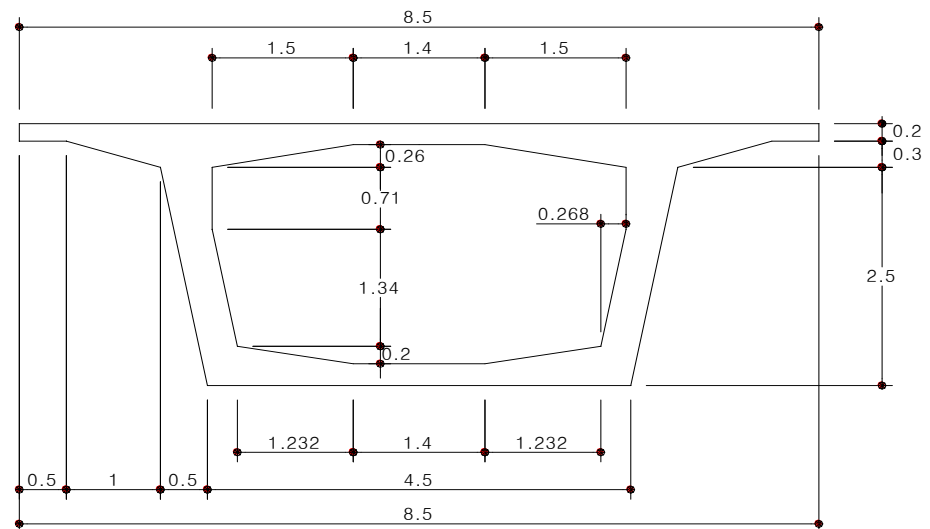


Figure 3. Cross Section

Materials & Strength

► Concrete

- 1) Specified Strength: $f_{cu} = 45MPa$
- 2) Modulus of Elasticity: $E_c = 3.0124 \times 10^4 MPa$

► PS Steel Tendons

- 1) Yield Strength: $f_{py} = 1580MPa$
- 2) Tensile Strength: $f_{pu} = 1860MPa$
- 3) Nominal Sectional Area: $A_p = 100cm^2$
- 4) Modulus of Elasticity: $E_p = 1.95 \times 10^5 MPa$
- 5) Initial Prestressing Force: $f_{pj} = 0.75f_{pu} = 1395MPa$
- 6) Anchorage Slip: $\Delta s = 6mm$
- 7) Coefficient of Curvature Friction : $\mu = 0.25 / rad$
- 8) Coefficient of Wobble Friction: $k = 0.0066 / m$

Loads

► Primary loads and special loads pertaining to the primary loads

- 1) Dead Loads
 - A. Reinforced Concrete: $24.52kN / m^2$
 - B. Asphalt Concrete: $22.56kN / m^2$
 - C. Barriers and safety fences
 - D. Prestress, creep, shrinkage
- 2) Live Loads
 - A. Vehicle Loads: *Types HA and HB Loading*
- 3) Differential settlements

: The worst combination of each pier settlement of 10mm

► Secondary loads

- 1) Temperature
 - A. For total deformation ($\pm 15^\circ$)
 - B. Temperature differential between top & bottom chords ($\pm 5^\circ$)
- 2) Wind

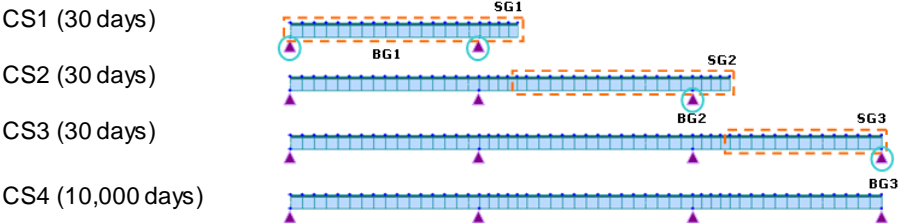
Composition of the Construction Stages

This figure below represents the entire construction stage process. Construction stages are generated excluding the erection of the shoring and temporary bents themselves, which have no effect on the structure.

Stage 1		i. Install temporary shoring (RP1, RP2 & P2) on temporary mass concrete foundation.
Stage 2		i. Install formwork (Stage 1). ii. Fabricate reinforcement, install sheath ducts & cast concrete (Stage 1). iii. Tension tendons (Stage 1).
Stage 3		i. Install formwork (Stage 2). ii. Fabricate reinforcement, install sheath ducts & cast concrete (Stage 2). iii. Tension tendons (Stage 2).
Stage 4		i. Remove temporary shoring (RP2 to P2) ii. Install temporary shoring (RP1 to A1)
Stage 5		i. Install formwork (Stage 3). ii. Fabricate reinforcement, install sheath ducts & cast concrete (Stage 3). iii. Tension tendons (Stage 3).

Figure 4. Construction Stage Chart

The following construction stages are reflected in the analysis.



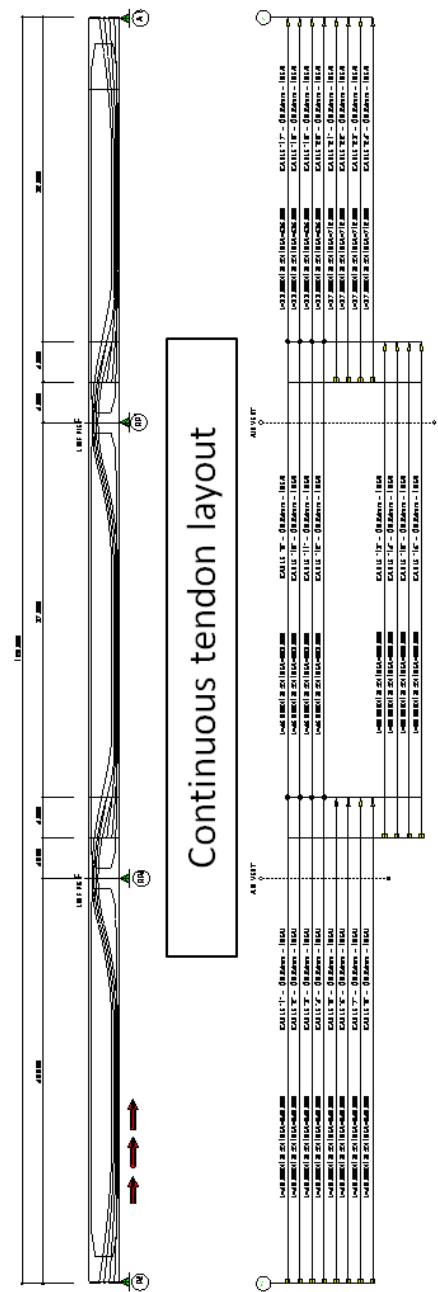




Figure 5. Tendon Placement Layout

Work Environment Settings



For FSM construction stage analysis, open a new file, ( **New Project**), and save as ( **Save**) '**FSM.mcb**'.

Select '**kN**' and '**m**' for the unit system. The unit system can be conveniently changed at any time later depending on your preferred types of input data.

 /  **New Project**
 /  **Save (FSM)**

Tools / **Unit System** 

Length> **m** ; Force>**kN** ↵

 The unit system can be changed by clicking "Unit Selection" () on the Status Bar at the bottom of the screen.

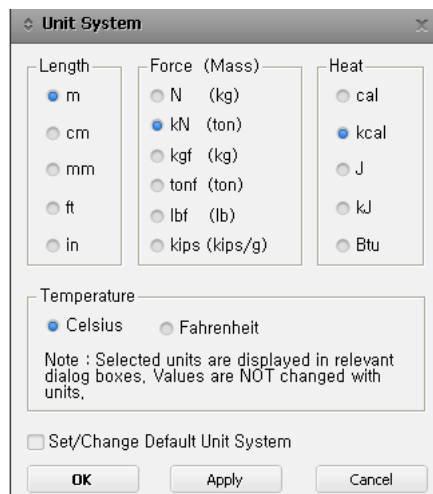








Figure 6. Unit System Setting

Definition of Properties

Definition of Materials

Define the material of the PSC box by selecting one from the built-in database. The material for tendons can be defined using the User Defined function.

Properties /  **Material Properties**
 Click  
 Type>**Concrete** ; Standard>**BS(RC)**
 DB>**C45** 
 Apply
 Name>**Tendon** ; Type>**User Defined**
 Modulus of Elasticity (**1.95e8**)
 Weight Density (**78.5**) 

 The tendon weight is automatically accounted for after grouting.

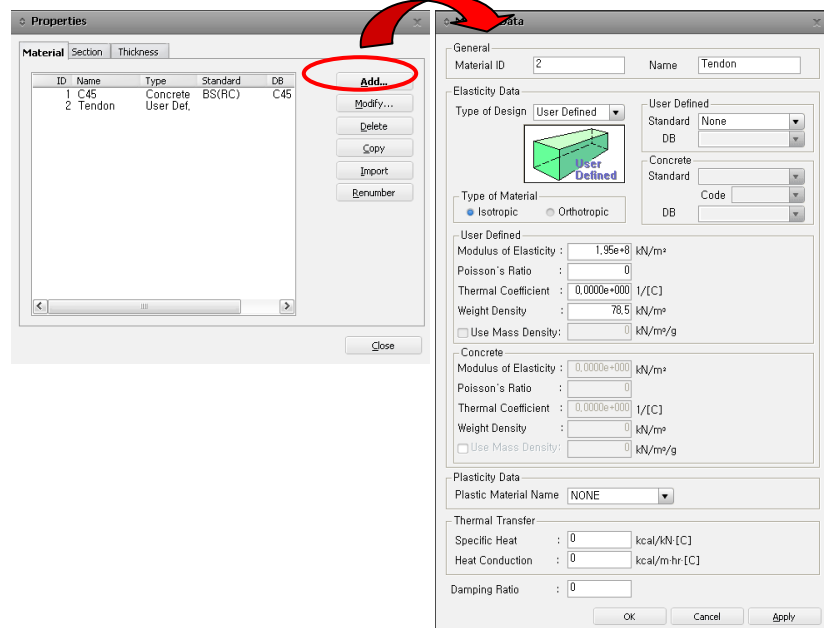


Figure 7. Material Data Input Dialog Box


Definition of Section

Refer to the cross section dimensions in Figure 8 to define the section of the PSC box.

Properties /  **Section Properties**
Click  ↵

PSC tab
Section ID **(1)** ; Name **(Span)**

PSC-1CELL, 2CELL
Joint On/Off>**JO1** (on), **J11** (on), **J13** (on), **J15** (on)
Web Thick> **for Shear t1** (on), **t2** (on), **t3** (on), **for Torsion(min)** (on)
Offset>**Center-Top**
Outer
HO1 **(0.2)** ; HO2 **(0.3)** ; HO2-1 **(0)** ; HO3 **(2.5)**
BO1 **(1.5)** ; BO1-1 **(0.5)** ; BO2 **(0.5)** ; BO3 **(2.25)**
Inner
HI1**(0.24)** ; HI2**(0.26)** ; HI2-1**(0)** ; HI3**(2.05)** ; HI3-1**(0.71)**
HI4 **(0.2)** ; HI4-1 **(0)** ; HI5 **(0.25)**
BI1**(2.2)** ; BI1-1**(0.7)** ; BI2-1**(2.2)** ; BI3**(1.932)** ; BI3-1**(0.7)**

Or click  to enter the input data in a table.

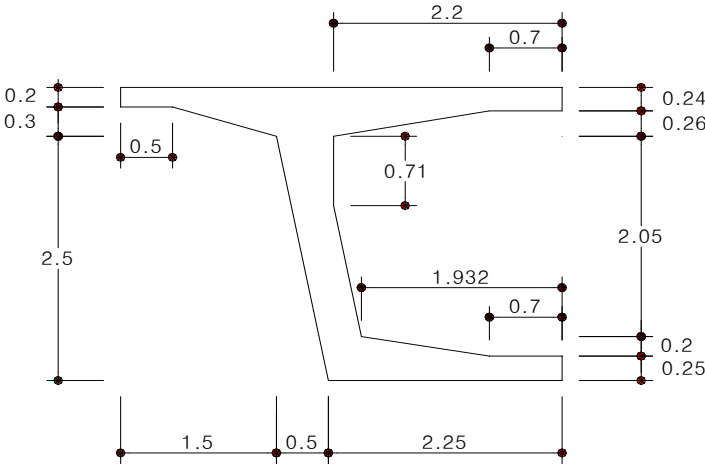


Figure 8. Input Data for the Cross Section

☞ Checking on
“Mesh Size for
Stiff Calc.”
enables us to
define a maximum
size of mesh,
which is used to
calculate the
section properties.

☞ “Consider Shear
Deformation”
accounts for shear
deformation.

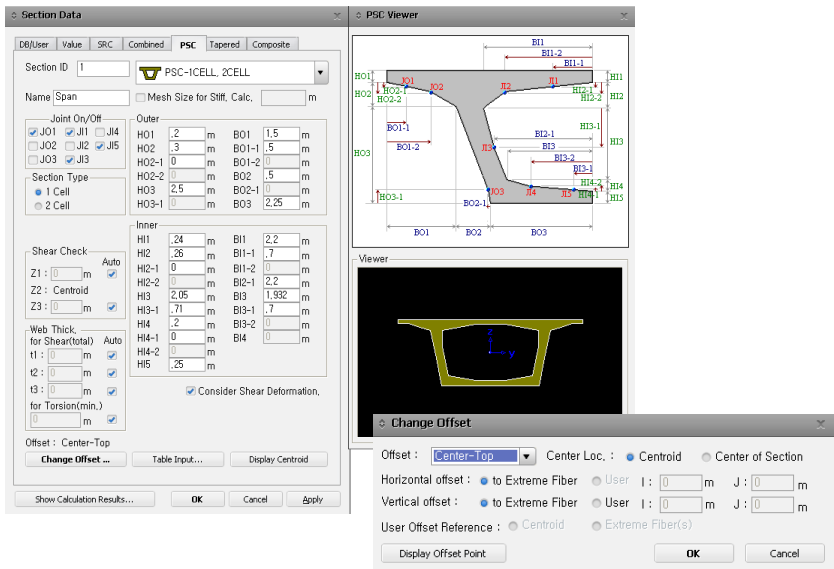


Figure 9. Section Input Dialog Box

Joint On/Off	H01	H02	H03	H01-1	H01-2	H02-1	H02-2	H03-1	H03-2	H01-1	H01-2	H02-1	H02-2	H03-1	H03-2
JO1															
JO2															
JO3															
JO4															
JO5															
Outer (m)															
H01	0.2000	0.3000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H02	0.3000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H02-1	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H02-2	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H03	2.5000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H03-1	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
Inner (m)															
H11	0.2400	0.2600	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H12	0.2600	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H12-1	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H12-2	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H13	2.0600	0.7100	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H13-1	0.7100	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H14	0.2000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H14-1	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H14-2	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
H15	0.2500														

Figure 10. Table Input (PSC)

Cross sectional dimensions can be entered via a table upon clicking **Table Input...** for the PSC section.

This is faster than directly entering the data in the dialog box for a large amount of dimensional data.

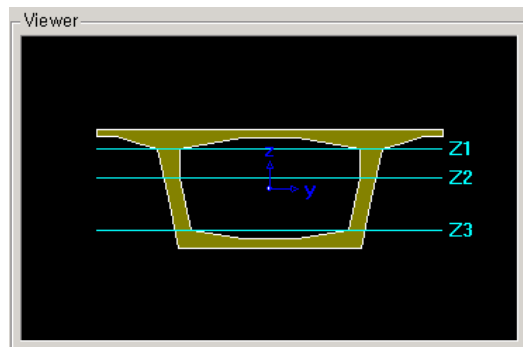
The table is compatible with Excel. Frequently used cross sectional dimensions can be saved to copy & paste later.

The table becomes compatible with Excel by entering “0” for Check Off (□) and “1” for Check on (☑).

Shear Check

Assign the locations for shear calculations on the PSC section. Numerical data can be entered manually, or if "Auto" is selected, shear calculations take place at the top and bottom of the web(s). The shear results are displayed in no. 5~10 of the Beam Stress (PSC).

Shear Check	
Z1 :	<input type="text" value="0"/> m <input checked="" type="checkbox"/> Auto
Z2 :	Centroid
Z3 :	<input type="text" value="0"/> m <input checked="" type="checkbox"/>



Web Thick.

Web Thick.	
for Shear(total) <input checked="" type="checkbox"/> Auto	
t1 :	<input type="text" value="0"/> m <input checked="" type="checkbox"/>
t2 :	<input type="text" value="0"/> m <input checked="" type="checkbox"/>
t3 :	<input type="text" value="0"/> m <input checked="" type="checkbox"/>
for Torsion(min.)	
<input type="text" value="0"/> m	<input checked="" type="checkbox"/>

for Shear(total)

Enter the thicknesses to be used for shear calculations at the locations defined for Shear Check at Z1 through Z3. Enter the sum of web thicknesses at a given location. Check on "Auto" for automatic calculations.


for Torsion(min.)

Enter a minimum thickness for torsion calculation.


Definition of Time-dependent Material Properties

Define the time-dependent properties of the concrete (creep coefficients shrinkage and strength).

Properties /  **Time Dependent Material/ Creep/Shrinkage**

Click  ; Name>**C45** ; Code>**CEB-FIP(1990)**
 Compressive strength of concrete at the age of 28 days (**45000**)
 Relative Humidity of ambient environment (40-99) (**70**)
 Notational size of member (**0.364**)
 Type of cement>**Normal or rapid hardening cement (N, R)**
 Age of concrete at the beginning of shrinkage (**3**) ↵

Properties /  **Time Dependent Material/ Comp. Strength**

Click  ; Name>**C45** ; Code>**CEB-FIP**
 Concrete Compressive Strength at 28 Days (**45000**)
 Type of cement>**N, R : 0.25** ↵

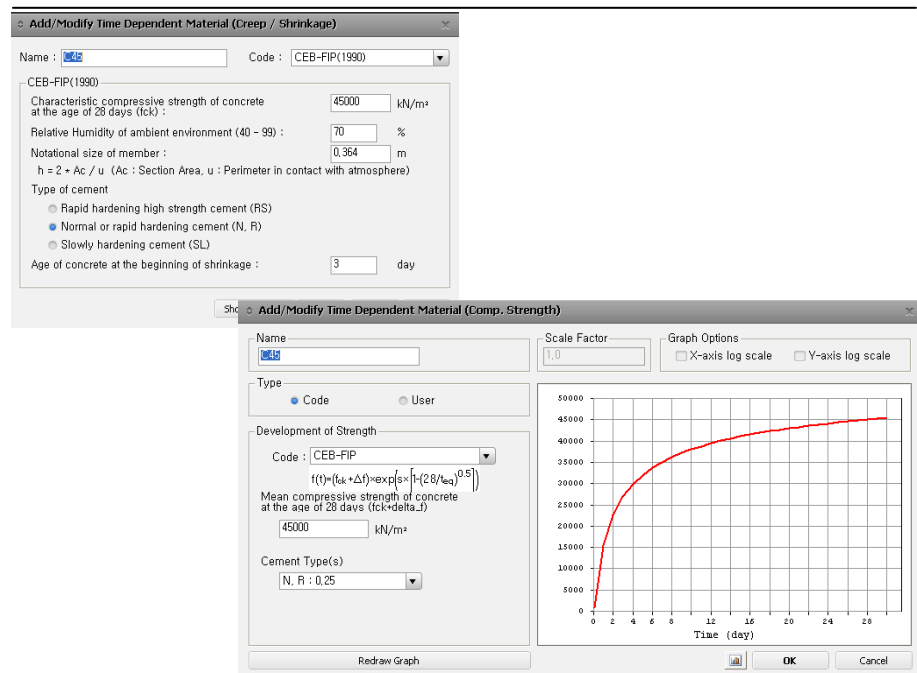




Figure 11. Time Dependent Material Data

Link the time dependent material properties to the material properties. The creep coefficients, shrinkage and concrete strength curves defined earlier need to be linked to the corresponding material property in order to carry out construction stage analysis reflecting their effects.

Properties / **Time Dependent** /  **Material Link**
 Time Dependent Material Type
 Creep/Shrinkage>**C45**
 Comp. Strength>**C45**
 Select Material for Assign>Materials>
1:C45 > Selected Materials 

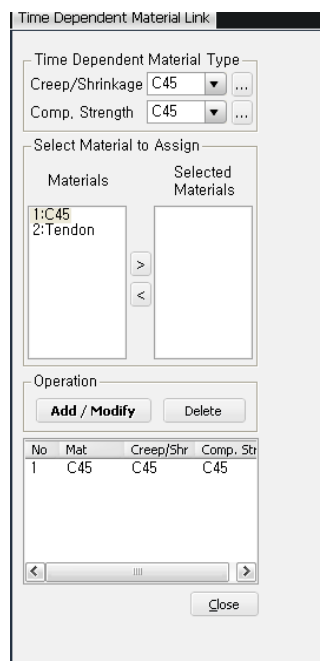



Figure 12. Linking Time Dependent Material Property to the Material Property.

Structural Modeling

Element Generation

Generate a girder using the “Extrude” function.

Node/Element /  **Create Nodes**

Coordinates (x, y, z) **(0, 0, 0)**  ↵

Node/Element/  **Extrude**

 **Select All**

Extrude Type>**Node** -> **Line Element**

Element Attribute


Element Type>**Beam**

Material>**1: C45**

Section>**1: Span**

Translation>Unequal Distance

Axis>**x**

Distances **(16@2.5, 5@2, 14@2.5, 5@2, 12@2.5)**  ↵

 **Zoom Fit**

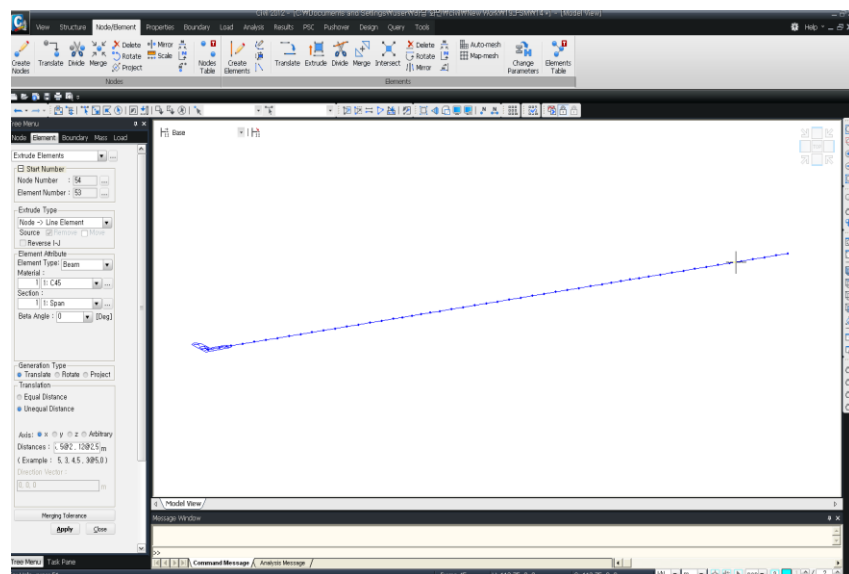











Figure 13. Girder Generation

Support Generation

Considering the spans (40+45+40), create nodes to which boundary conditions will be assigned.

Since the depth of the girder is 3m, and the distance between the bearings is 3m with the working point being Center-Top, the supports are created at Z=-3m & Y=±1.5m.

Node/Element /  **Create Nodes**
 Start Node Number  ↵
 Node Numbering Option > **User-Defined Number**
 Newly Created Number (61)  ↵
 Coordinates (x, y, z) (0, 1.5, -3)  
 Copy
 Number of Times (1)
 Distances (dx, dy, dz) (0, -3, 0)  ↵

Node/Element /  **Translate Node**
 **Select Recent Entities**
 Mode > **Copy**
 Translation
 Unequal Distance
 Axis > **x**
 Distance (40, 45, 40)  ↵

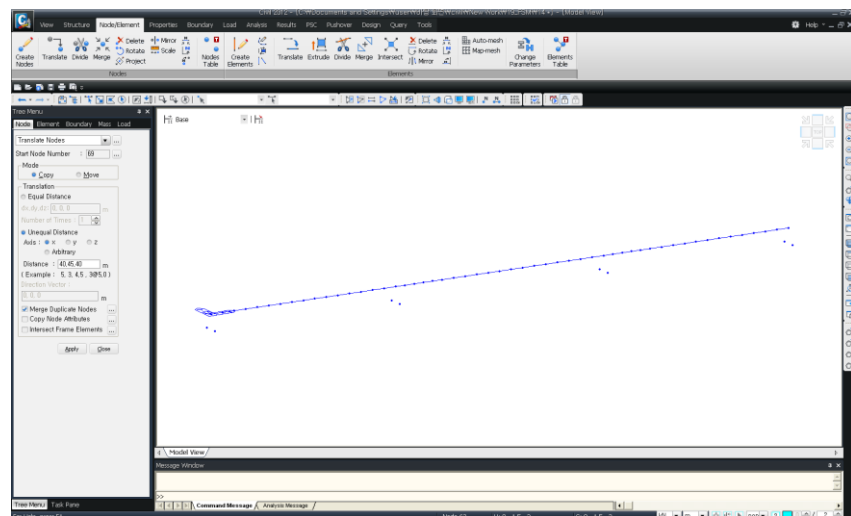




Figure 14. Generation of Support Nodes


Group Definition


Refer to “Construction Stage Configuration” on Figure 4 for the list of the groups to be defined.

🔊 Tendon Group is not used for composing the construction stages, but is defined to check the results for each group.

Structure / Group /  **Structure**
Name (SG) ; Suffix (1to3) ↵

Structure / Group / B/L/T /  **Define Boundary Group**
Name (BG) ; Suffix (1to3) ↵

Structure / Group / B/L/T /  **Define Load Group**
Name (Dead) ; ↵ ;
Name (Superimposed Dead) ; ↵
Name (PS) ; Suffix (1to3) ; ↵
Name (Diaphragm) ; Suffix (1to3) ; ↵

Structure / Group / B/L/T /  **Define Tendon Group** 🔊
Name (A) ; Suffix (1to4) ↵
Name (B) ; Suffix (1to4) ↵
Name (C) ; Suffix (1to4) ↵

🔊 Refer to the name assignment of the “Tendon Profile” to see the items of the Tendon Group.

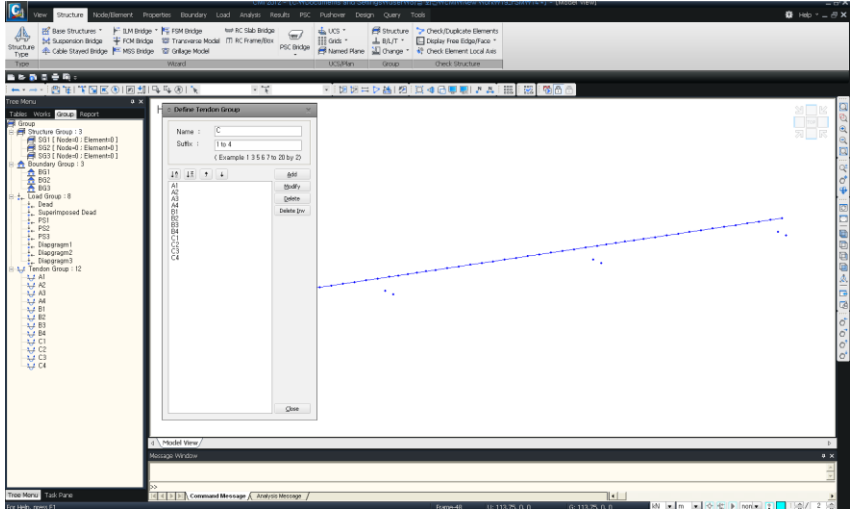


Figure 15. Group Generation


Structure Group Assignment

Assign the elements, which will be activated at each stage, to SG1~3 respectively. Assign the elements to Structure Group by using “Drag & Drop,” or by right-clicking and selecting “Assign”.

 **Elements Numbers ;**  **Front View**

Group Tab in the Tree Menu

Type the numbers of nodes and elements as below

 61to64   1to20 

Select Nodes : **61to64** & Elements : **1to20**

Structure Group > SG1 **Drag & Drop** or (Context Menu) **Assign**

Select Nodes : **65to66** & Elements : **21to39**

Structure Group > SG2 **Drag & Drop** or (Context Menu) **Assign**

Select Nodes : **67to68** & Elements : **40to52**

Structure Group > SG2 **Drag & Drop** or (Context Menu) **Assign**

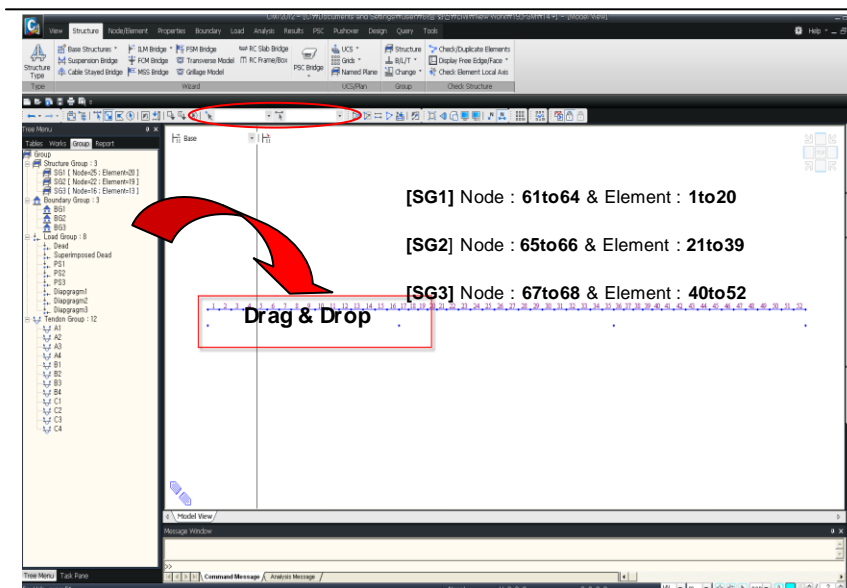


Figure 16. Structure Group Assignment

Boundary Conditions Input

Rigid Links

Considering the centroid of the cross section of the PSC Box, rigid links are connected to the supports.

Iso View

Boundary /  Elastic Link

Boundary Group>**BG1**

Link Type>**Rigid Type**

2Nodes **(1,61) ; (1,62) ; (18,63) ; (18,64)** ↵

Boundary Group>**BG2**

2Nodes **(37,65) ; (37,66)** ↵

Boundary Group>**BG3**

2Nodes **(53,67) ; (53,68)** ↵

Turn on the node number if necessary when picking up nodes

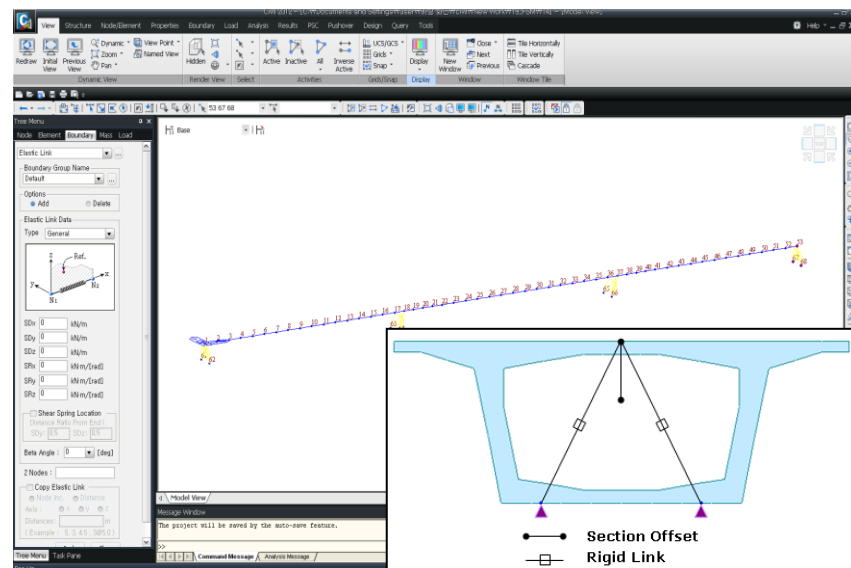


Figure 17. Rigid Links

Supports Input

Considering the construction stages, the supports are defined as below.

Top View ; **Redraw**

Boundary / **Define Supports**
Boundary Group>**BG1**

- Select Single** (Node : **61**) ; Support Type>**Dy** (on), **Dz**(on) ↵
- Select Single** (Node : **62**) ; Support Type>**Dz**(on) ↵
- Select Single** (Node : **63**) ; Support Type>**Dx**(on), **Dy**(on), **Dz**(on) ↵
- Select Single** (Node : **64**) ; Support Type>**Dx**(on), **Dz**(on) ↵

Boundary Group>**BG2**

- Select Single** (Node : **65**) ; Support Type>**Dy**(on), **Dz**(on) ↵
- Select Single** (Node : **66**) ; Support Type>**Dz**(on) ↵

Boundary Group>**BG3**

- Select Single** (Node : **67**) ; Support Type>**Dy**(on), **Dz**(on) ↵
- Select Single** (Node : **68**) ; Support Type>**Dz**(on) ↵

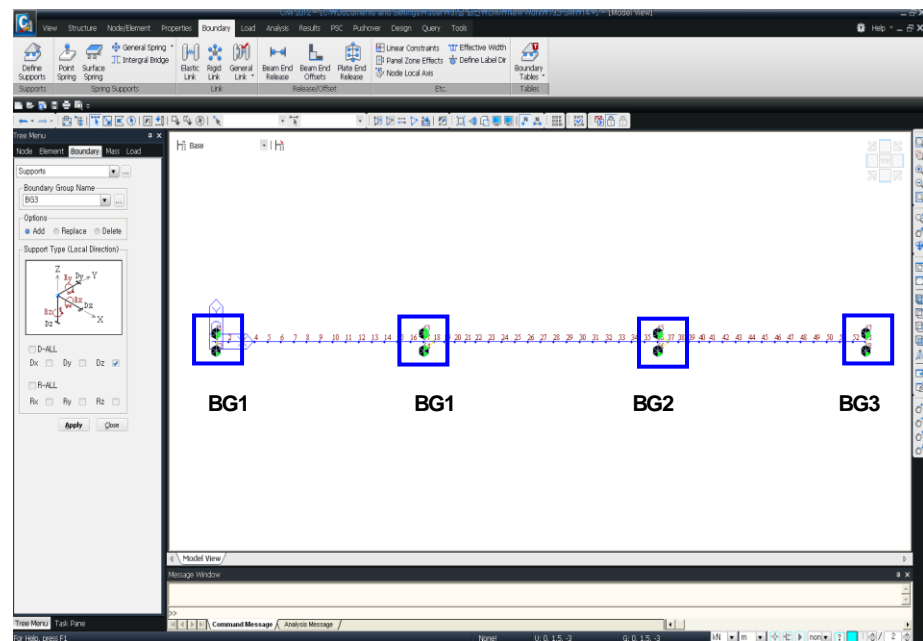


Figure 18. Boundary Condition Input

Construction Stage Loads Input

Define Load Conditions

Define load cases for analysis.

We take the time to define the load “Type” then we can take advantage of the ability to automatically generate load combinations using the "Auto Generate" function. Using these Types of load case we may generate the load combinations after application of the load factors as per the design standard.

 **Redraw**

Load /  **Static Load Cases**

Name (Self Weight)	; Type>Construction Stage Load (CS) ↵
Name (Non-Structure Dead)	; Type>Construction Stage Load (CS) ↵
Name (Prestress)	; Type>Construction Stage Load (CS) ↵
Name (Superimposed)	; Type>Construction Stage Load (CS) ↵
Name (Wind)	; Type>Wind Load on Structure (W) ↵
Name (Temperature (+))	; Type>Temperature (T) ↵
Name (Temperature (-))	; Type>Temperature (T) ↵
Name (Top-Bot Temp Diff(+))	; Type>Temperature Gradient (TPG) ↵
Name (Top-Bot Temp Diff(-))	; Type>Temperature Gradient (TPG) ↵

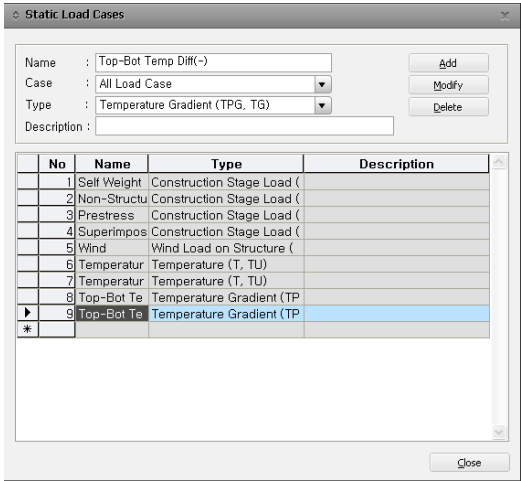




Figure 19. Load Cases Definition

Self Weight

Enter the self weight.

Define the structure's self weight and activate it at the first construction stage. Then the self weights of the elements activated in the subsequent construction stages will automatically be applied.

Load /  **Self Weight**
 Load Case Name>**Self Weight**
 Load Group Name>**Dead**
 Self Weight Factor>**Z (-1)**  ↵

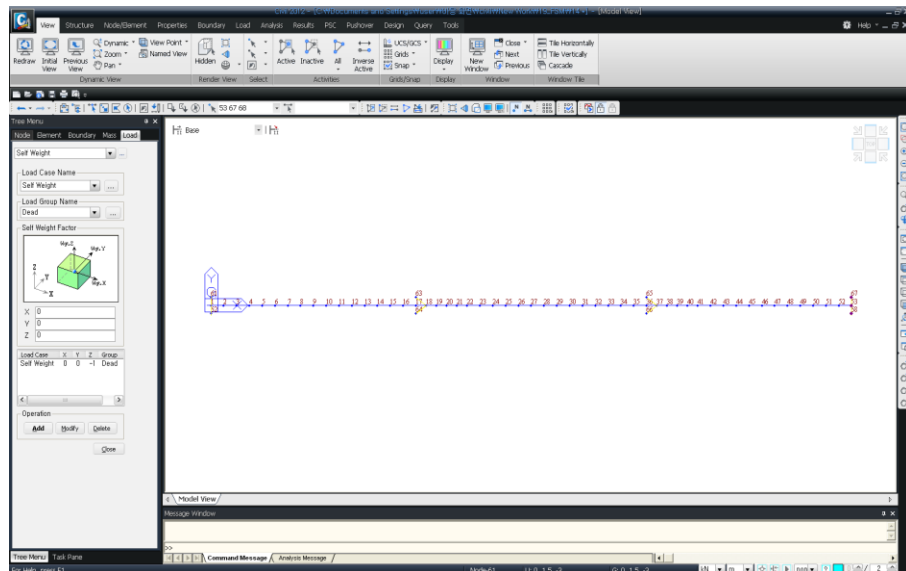


Figure 20. Self Weight Input

Dead Load

Enter diaphragms and construction joint blocks, as loads as they have not been reflected in the model.

Front View

Load / Element Beam Loads

1 52 **Select Elements by Identifying (1, 52)** ↵

Load Case Name **Non-Structure Dead**

Direction **Global Z**

Relative

x1(0) ; x2(1) ; w(-220.34) **Apply** ↵

19to21 38to40 **Select Elements by Identifying (19to21, 38to40)** ↵

Relative

x1(0) ; x2(1) ; w(-63.0) **Apply** ↵

16 35 **Select Elements by Identifying (16, 35)** ↵

Absolute

x1(1.5) ; x2(2.5) ; w(-220.34) **Apply** ↵

17 36 **Select Elements by Identifying (17, 36)** ↵

Absolute

x1(0) ; x2(1) ; w(-220.34) **Apply** ↵

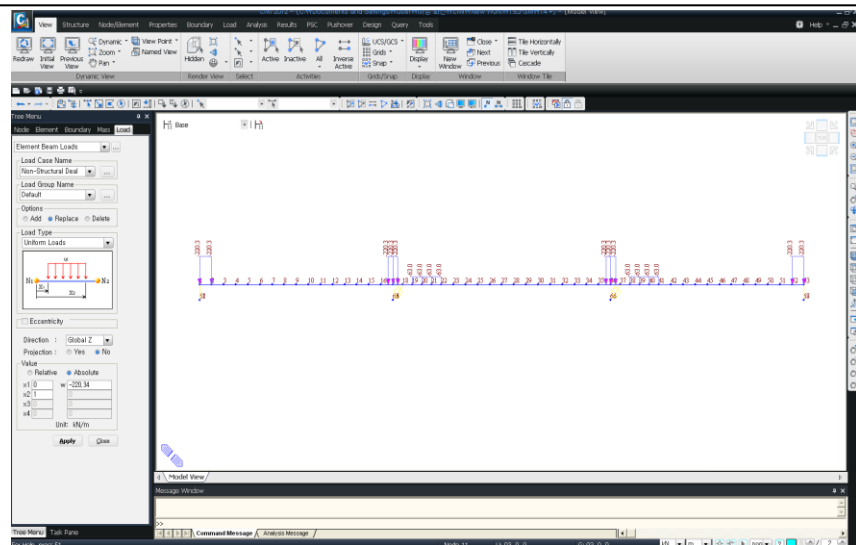


Figure 21. Miscellaneous Dead Loads

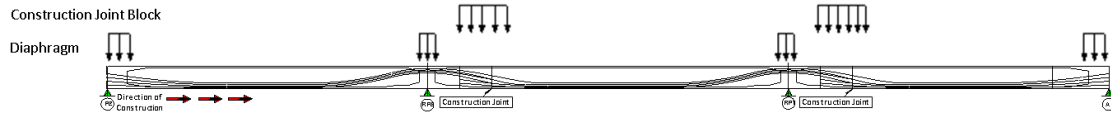


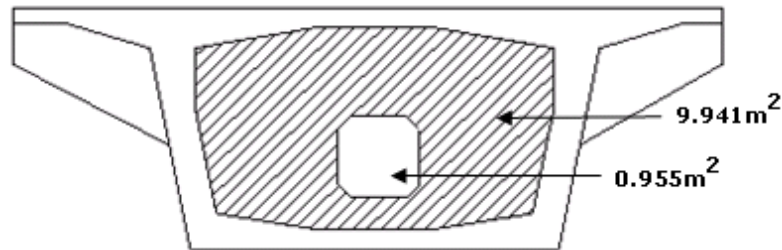
Figure 22. Dead Load Layout

The diaphragms at the supports and the construction joint blocks have not been considered as structural elements in this longitudinal analysis and are thus treated as loads. Their cross sectional areas are calculated and converted into Beam Load over the corresponding lengths. Other additional dead loads may exist, but are ignored in this Tutorial.

Diaphragm (End: 2m, Intermediate Support: 2.5m)

$$Area = 9.941m^2 - 0.955m^2 = 8.986m^2$$

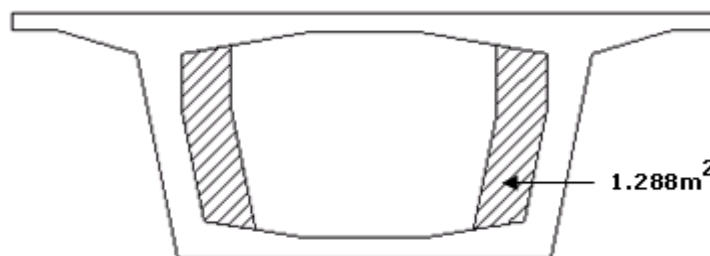
$$P = 8.986m^2 \times 24.52kN/m^3 = 220.34kN/m$$



Construction Joint Block

$$Area = 1.288m^2 \times 2EA = 2.576m^2$$

$$P = 2.576m^2 \times 24.52kN/m^3 = 63kN/m$$



We need to assign the loads to Load Groups and activate the Load Groups in the corresponding construction stages.

Because the magnitudes of the Beam Loads are the same, setting the Load Group to

Default is convenient for input. We will now see how to modify the Load Group using the Table Tab.

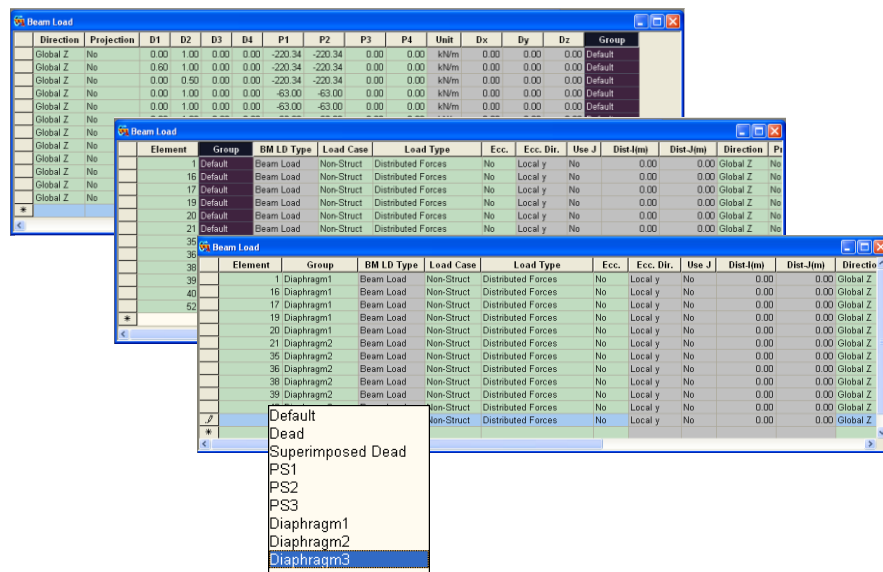
By selecting the desired columns, we can adjust the locations in Beam Load Table. The row column containing the Group information is located at the end of the Table. For convenience, we will select the entire column, and move it next to the Element numbers.

Assign Load Group: Diaphragm1 to 3 to the loads in order to activate them in Stages 1 through 3.

Load / Load Tables / Static Load / **Beam Loads** **Assign**

Element 1~20> **Diaphragm1**
Element 21~39> **Diaphragm2**
Element 40~52> **Diaphragm3**

Note that the **Group** column is found at the last column in the table as shown in the first figure of the three figures below. In the second figure, the **Group** column was relocated to the front for convenience. The third figure depicts how Diaphragm is applied to the elements.



Direction	Projection	D1	D2	D3	D4	P1	P2	P3	P4	Unit	Dx	Dy	Dz	Group
Global Z	No	0.00	1.00	0.00	0.00	-220.34	-220.34	0.00	0.00	kN/m	0.00	0.00	0.00	Default
Global Z	No	0.60	1.00	0.00	0.00	-220.34	-220.34	0.00	0.00	kN/m	0.00	0.00	0.00	Default
Global Z	No	0.00	0.50	0.00	0.00	-220.34	-220.34	0.00	0.00	kN/m	0.00	0.00	0.00	Default
Global Z	No	0.00	1.00	0.00	0.00	-63.00	-63.00	0.00	0.00	kN/m	0.00	0.00	0.00	Default
Global Z	No	0.00	1.00	0.00	0.00	-63.00	-63.00	0.00	0.00	kN/m	0.00	0.00	0.00	Default


Element	Group	BM LD Type	Load Case	Load Type	Ecc.	Ecc. Dir.	Use J	Dist-Im	Dist-Jm	Direction	P1
1	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
16	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
17	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
19	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
20	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
21	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No


Element	Group	BM LD Type	Load Case	Load Type	Ecc.	Ecc. Dir.	Use J	Dist-Im	Dist-Jm	Direction	P1
1	Diaphragm1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
16	Diaphragm1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
17	Diaphragm1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
19	Diaphragm1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
20	Diaphragm1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
21	Diaphragm2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
36	Diaphragm2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
36	Diaphragm2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
38	Diaphragm2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
39	Diaphragm2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	Default	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	Dead	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	Superimposed Dead	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	PS1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	PS2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	PS3	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	Diaphragm1	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	Diaphragm2	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No
	Diaphragm3	Beam Load	Non-Struct	Distributed Forces	No	Local y	No	0.00	0.00	Global Z	No

Figure 23. Changing Load Group using Table

Tendon Prestress Load

Define the properties of the Tendon related to the material, strength, losses. etc

Load / Temp./Prestress /  **Tendon Property**

Click  ↵


Tendon Name (**Tendon**)

Tendon Type> **Internal(Post-Tension)** ; Material> **2: Tendon**

Total Tendon Area> **0.0016112**

or Strand Diameter> **12.9mm(1x3)** ; Number of Strands> **19**


Duct Diameter (**0.1**)


Relaxation Coefficient> **CEB-FIP(2.5%)** 


Ultimate Strength> (**1860000**) ; Yield Strength> (**1580000**)

Curvature Friction Factor> (**0.25**) ; Wobble Friction Factor> (**0.0066**)

Anchorage Slip(Draw in)> Begin(**0.006**) , End(**0.006**)

Bond Type> **Bonded** 

 **Relaxation**
Coefficient can be
defined by selecting
Magura equation,
JTG04 or CEB-FIP
Code.

 If “Unbonded”
is selected, the
section stiffness is
calculated on the
basis of the net
cross section.
“Bonded” reflects
the composite
stiffness reflecting
the tendons.

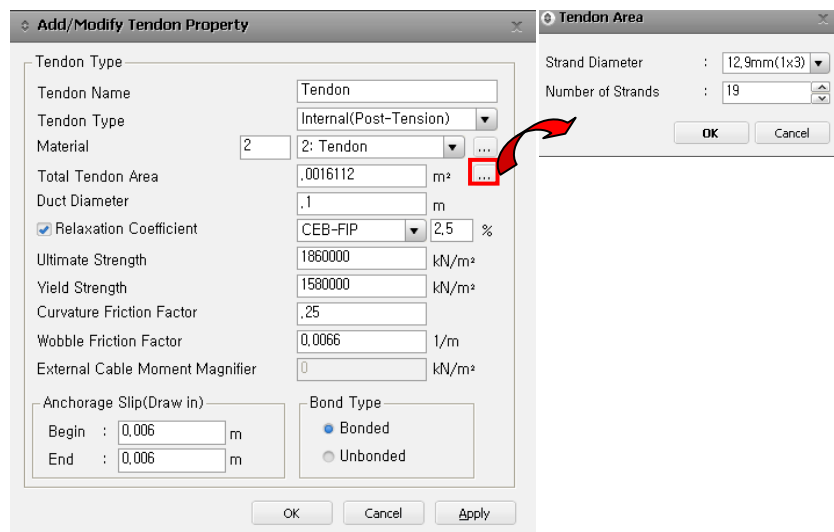





Figure 24. Tendon Property Dialog


The Tendon Profile can be defined in many ways such as defining the inflection points, but this example uses a common approach often used in practice, using the Tendon ordinates from drawings.

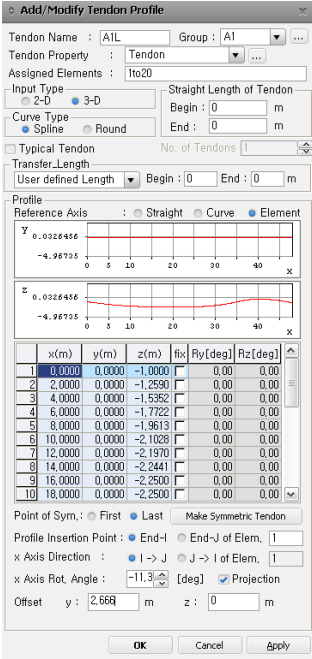
Referring to the values in the attached Excel file (*TD profile.xls*), prepared on the basis of the tendon drawings the ordinates of the tendon at every 2m are pasted into the software.

Load / Temp./Prestress /  **Tendon Profile**
 Tendon Name (**A1L**) ; Group (**A1**)
 Tendon Property> **Tendon** ; Assigned Elements (**1to20**)
 Input Type> **3-D** ; Curve Type> **Spline**
 Profile
 1> x (0), y (0), z (-1)
 2> x (2), y (0), z (-1.2590)
 ...
 25> x (48), y (0), z (-1.25)
 Profile Insertion Point> **End-I of Elem.1**
 x Axis Direction> **I->J of Elem.1** ; x Axis Rot. Angle (**-11.3**)
 Offset y : (**2.666**)

 Copy & Paste the values from the Excel file to enter the Profile. We may also copy the Profile after creating an MCT file.

 Transfer Length may be specified to consider the unstressed length of the anchorage.

 Checking on “Typical Tendon” and entering the number of tendons can be used to represent a number of tendons of the same profile. This is also handy when preliminary analysis is



	x(m)	y(m)	z(m)	fix	Ry[deg]	Rz[deg]
1	0.0000	0.0000	-1.0000		0.00	0.00
2	2.0000	0.0000	-1.2590		0.00	0.00
3	4.0000	0.0000	-1.5362		0.00	0.00
4	6.0000	0.0000	-1.7722		0.00	0.00
5	8.0000	0.0000	-1.9613		0.00	0.00
6	10.0000	0.0000	-2.1028		0.00	0.00
7	12.0000	0.0000	-2.1970		0.00	0.00
8	14.0000	0.0000	-2.2441		0.00	0.00
9	16.0000	0.0000	-2.2500		0.00	0.00
10	18.0000	0.0000	-2.2500		0.00	0.00

Figure 25. Tendon Profile Input Dialog

From the tendon profile drawings, x-z coordinates are obtained at every 2m. The result (TD Profile.xls) contains the values as if the tendons were placed in the centroidal 2-D plane, each side. We need to translate the layout using y-Offset and rotate the layout using x-Rotation to properly position them in the webs of the PSC section.

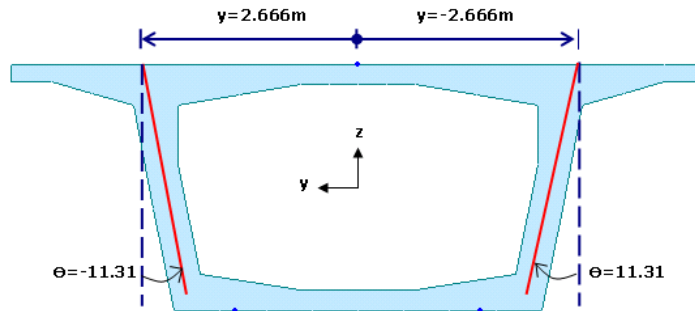
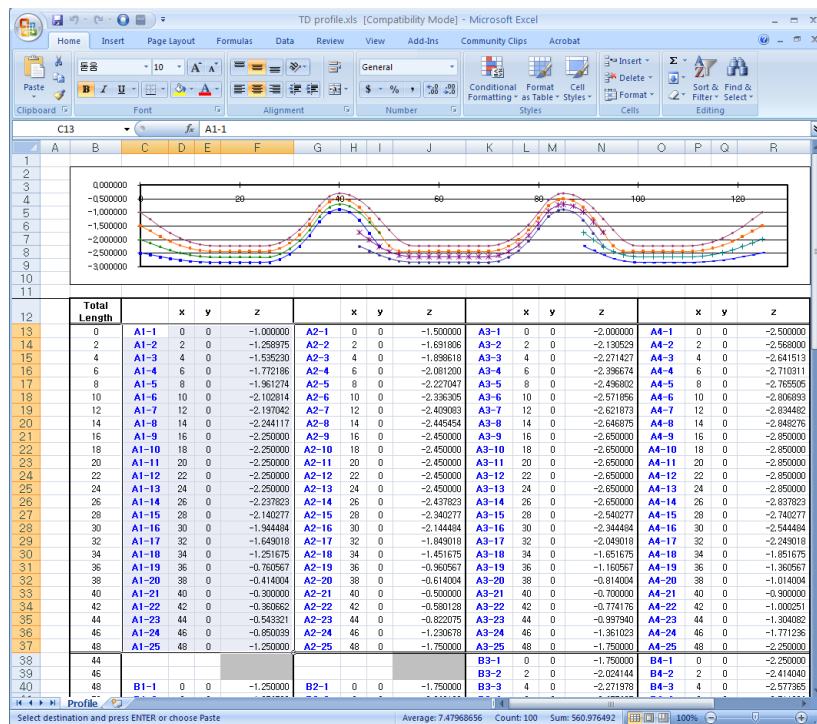


Figure 26. 3-Dimensional Tendon Profile Input

Copy and paste the values of x, y and z from the Excel file as below, and position the tendons in the webs by y-Offset and x-Rotation depending on the "left" or "right" tendon.



The Name and the Assigned Elements for all Tendon Profiles are as follows:

Ex) A1L → X coordinate (A, B, C), Z coordinate (1, 2, 3, 4), Y coordinate (Left, Right)

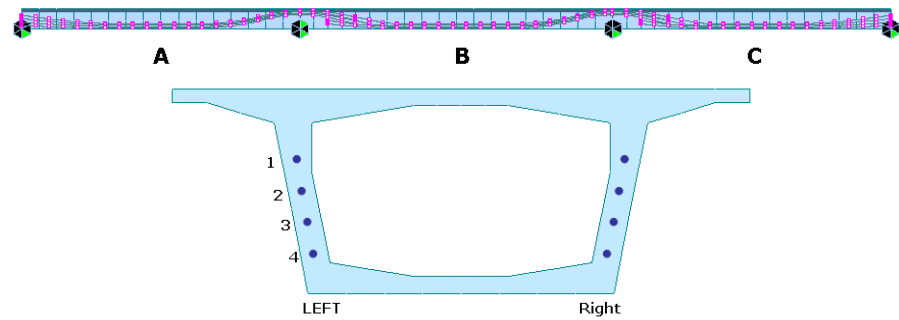


Figure 27. Name Assignment for Tendon Profile

Tendon Profile	Assigned Element	Tendon Profile	Assigned Element
A1, A2	1 ~ 20	A3, A4	1 ~ 20
B1, B2	21 ~ 39	B3, B4	19 ~ 39
C1, C2	40 ~ 52	C3, C4	38 ~ 52

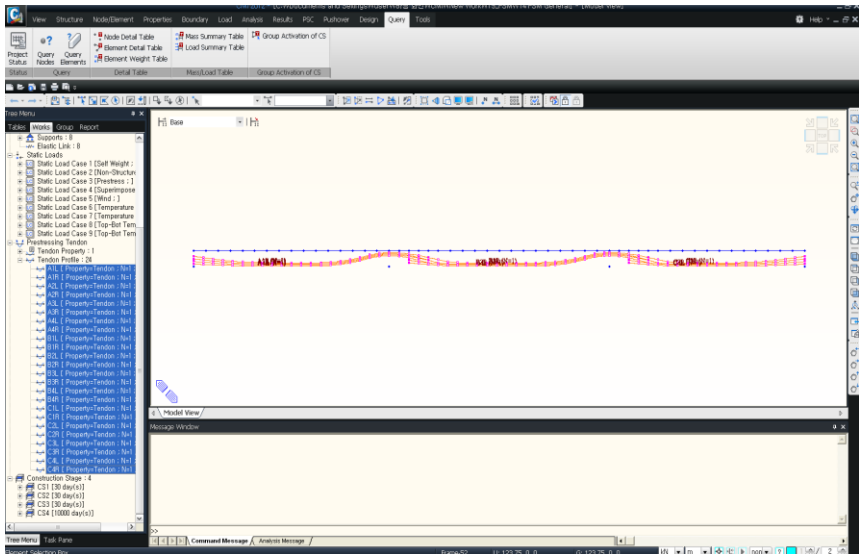


Figure 28. Result of Tendon Profile Input

After defining all the Tendon Profiles, assign the Load Groups (PS1~3) and then apply prestress loads so that the defined Tendon Profiles can be applied to each construction stage.

🔊 Prestress is applied one stage after the stage at which the load is entered.

Load / Temp.Prestress / **Tendon Prestress Loads**

Load Case Name> **Prestress**

Load Group Name> **PS1**

Select Tendon for Loading

Tendon> **A1L~A4R** >

Stress Value

Begin (**1395000**) ; End (**0**)

Grouting: after (**1**) Stage  ↵


Load Group Name> **PS2**

Select Tendon for Loading

Selected> **A1L~A4R** < ; Tendon> **B1L~B4R** >  ↵

Load Group Name> **PS3**

Select Tendon for Loading

Selected> **B1L~B4R** < ; Tendon> **C1L~C4R** >  ↵

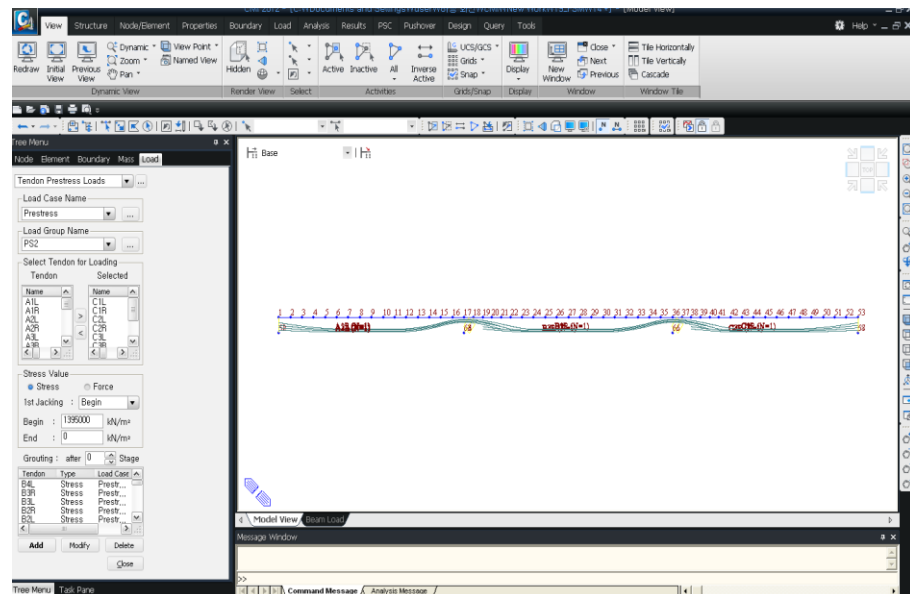


Figure 29. Loading Tendon Prestress

Superimposed Dead Loads

Superimposed Dead Loads are applied as Beam Load onto the superstructure.

Barriers	$(0.3075m^2 + 0.4975m^2) \times 24.52kN/m^3$	$19.74kN/m$
Safety Fences		$1kN/m$
Asphalt concrete pavement	$7.5m \times 8cm \times 22.56kN/m^3$	$13.5kN/m$
Noise barriers		$1.52kN/m$
Total		$35.796kN/m$

Load / Static Loads / Beam Loads /  **Element**



 **Select All**

Load Case Name>**Superimposed**

Load Group Name> **Superimposed dead**

Load Type>**Uniform Loads**

Value

Relative ; x1(0) ; x2 (1) ; w(-35.796)  

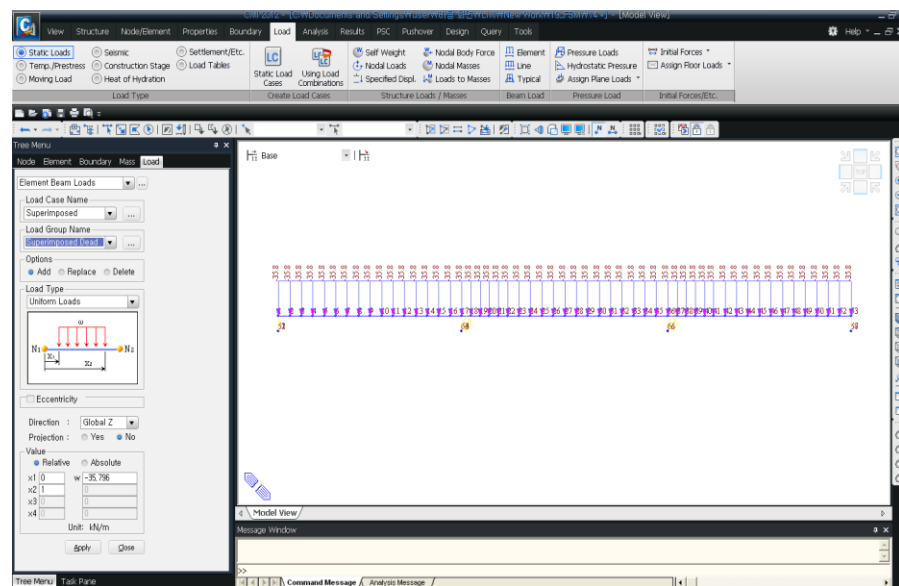


Figure 30. Loading Superimposed dead Loads

Loading Input on the Completed Structure

Wind Loading

wind loading of 3 kN/m^2

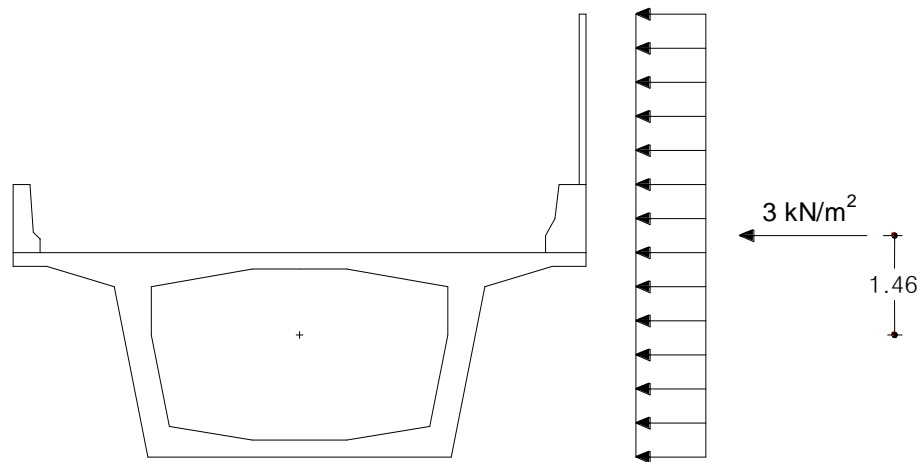


Figure 31. Wind Load Distribution


Total Height = Section Depth + Barriers + Noise barriers = $3 + 1 + 2.5 = 6.5 \text{ m}$

Wind Pressure = 3 kN/m^2

Wind Load = $6.5 \text{ m} \times 3 \text{ kN/m}^2 = 19.5 \text{ kN/m}$ (Horizontal Load)

= $19.5 \text{ kN/m} \times -1.46 \text{ m} = -28.47 \text{ kN} \cdot \text{m/m}$ (Eccentricity Moment)

Enter the wind loads.

Load / Static Loads/ Beam Loads /  **Element**

 **Select All**  **Iso View**

Load Case Name>**Wind**



Load Group Name>**Default** 

Load Type>**Uniform Loads**

Direction>**Global Y**

Value

Relative

x1(0) ; x2 (1) ; w(19.5)  

 **Select All**


Load Type>**Uniform Moments/Torsion**

Direction>**Global X**

Value

Relative

x1(0) ; x2 (1) ; w(-28.47)  

 **Loading**
pertaining to the
Load Groups,
which are not
activated during
the construction
stages, are
loaded in
PostCS.

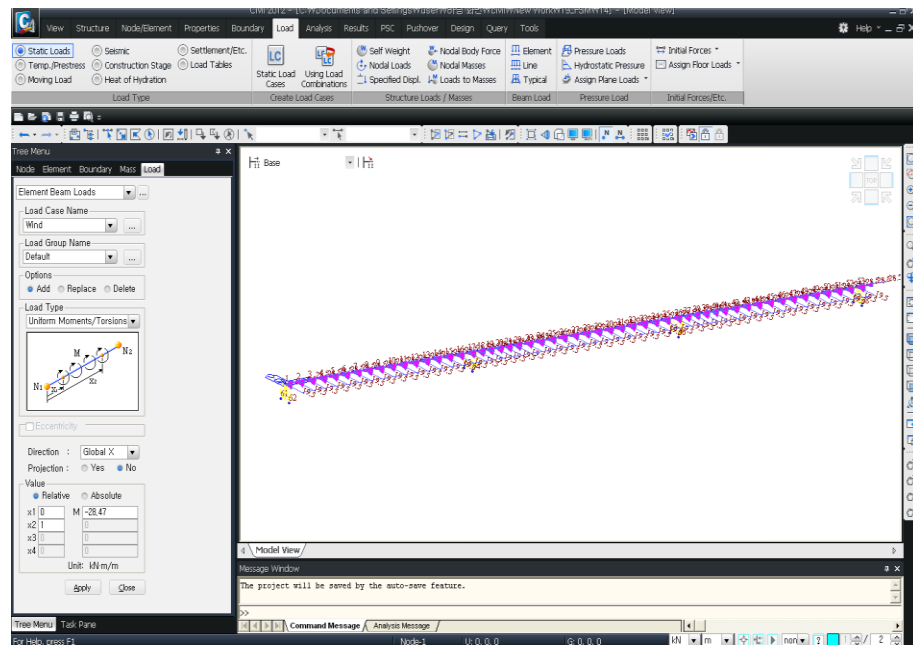



Figure 32. Wind Loading Input

Temperature

Specify the temperature loading acting on the entire structure.

The System Temperature function allows us to specify strain, $\varepsilon_t = \alpha(T_2 - T_1)$, over the entire structure as temperature loads.



Load / Temp./Prestress / Temperature Loads /  **System Temp.**



Redraw



Load Case Name>**Temperature (+)**

Load Group Name>**Default**

Temperature > Final Temperature **(15)**  

Load Case Name>**Temperature (-)**

Load Group Name>**Default**

Temperature > Final Temperature **(-15)**  

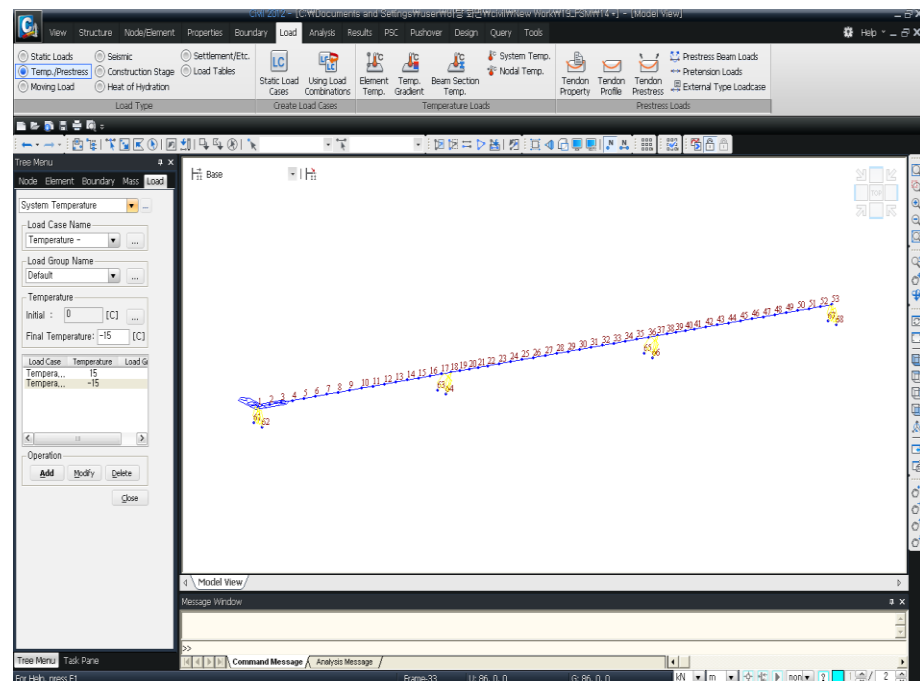


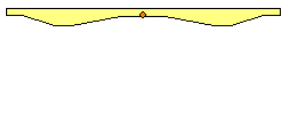
Figure 33. Temperature Loading Input

Specify the differential temperature between the top and bottom chords.

The Beam Section Temperature function generates a temperature differential between top and bottom chords on a part of a rectangle. Since PSC sections are not rectangular sections, they need to be converted into equivalent rectangular sections to be able to specify temperature differential loads.

Where temperature differentials exist as shown below, the parts experiencing the temperature differentials are converted into a rectangle defined by dotted lines having the same area and centroid.



Result List	
Section	test
Base Material	
	
	Value [Length-Unit:m]
Area	2.896000
SAx	2.033214
SAy	0.502343
Ixx	0.044395
Iyy	16.593738
Ixy	-0.000000
J	0.142504
(+)Cx	4.250000
(-)Cx	4.250000
(+)Cy	0.187023
(-)Cy	0.312977
List Order	Creation
<div>Modify</div> <div>Close</div>	

Beam Section Temperature can be defined as either General Type or PSC Type. General Type assumes the section as a rectangle. When PSC Type is specified, the sections defined as PSC Type in defining Section Data are automatically converted into rectangles and loaded on the parts experiencing temperature differentials.


Although the Beam Section is defined as PSC Type in this example, which results in a simple input process for loading for a temperature differential between the top and bottom chords, input is carried out as General Type after converting into a rectangle.

Figure 34 shows the calculations for cross sectional area and centroid of the top part of the PSC Box section using SPC (Section Property Calculator). The instruction for using SPC is separately documented in user's manual.

Figure 34. Section Properties calculated by SPC

Using the above calculation results in conversion into an equivalent rectangle, which will be loaded, as follows:



$$\begin{aligned} \text{Area} &= 2.896 \text{ m}^2 \\ H &= 2 \times 0.312977 \text{ m} = 0.625954 \text{ m} \\ \therefore B &= \frac{\text{Area}}{H} = \frac{2.896}{0.625954} = 4.626 \text{ m} \end{aligned}$$

Load / Temp./Prestress / Temperature Loads /  **Beam Section Temp.**

Load Case Name> **Top-Bot Temp Diff (+)**

Load Group Name> **Default**



Direction > **Local-z** ; Ref. Position > **Centroid**

B (4.626) ; H1 (0.71) ; H2 (1.336) ; T1 (5) ; T2 (5)  

 **Select All**  

Delete the defined Section Temperatures (select ① and delete)

Load Case Name> **Top-Bot Temp Diff (-)**

B (4.626) ; H1 (0.71) ; H2 (1.336) ; T1 (-5) ; T2 (-5)  

 **Select All**  

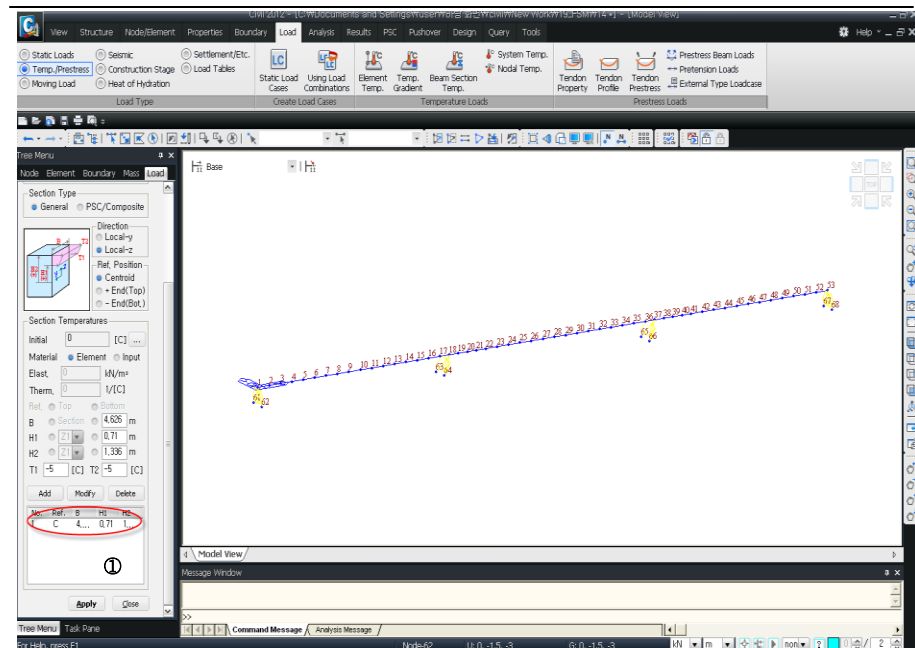


Figure 35. Input for Temperature Differential between Top & Bottom Chords

Live Load

The sequence of defining the live load is as follows:

Select a Code defining live load: Define Moving Load Code
 Define lanes: Traffic Line Lanes
 Define vehicles: Vehicles
 Define live load cases: Moving Load Cases

► Select a Code, which specifies live load

The input process and the parameters are tailored to the selected Code.

Load / Moving Load Load Type /  **Moving Load Code**

Moving Load Code>**BS**

► Define traffic lanes

Eccentric and symmetrical loading can be considered for the transverse position of traffic lanes. In this tutorial, we specify only a symmetrical loading case as described below.

The eccentricity is positive (+) if the traffic lane (center) is on the right side of the elements in the direction of traffic, and vice versa.

🔊 Since this example bridge is straight and symmetrical, only the wind loading in the +Y direction has been applied. For the worst condition, only the eccentric live load in the +Y direction is entered.

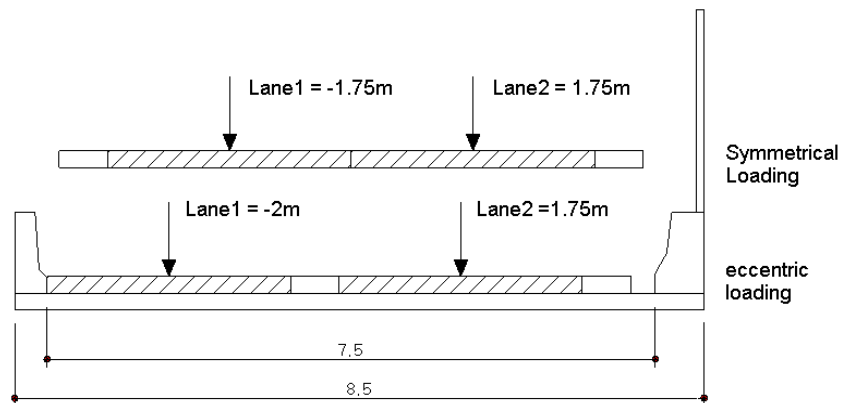


Figure 36. Traffic Lanes & Eccentricities

Refer to the Figure 36 for the traffic lanes and eccentricities to define 2 traffic lanes.

Top View

Load / Moving Load / **Traffic Line Lanes**

Click **Add** ↵

Lane Name (**Lane 1 left**)

Traffic Lane Properties

Eccentricity (**-1.75**) ; Wheel Spacing (**1**) ; Lane Width(**3.5**)

Vehicular Load Distribution > **Lane Element**

Selection by > **2 Points ((0, 0, 0)(125, 0, 0))**

Click **OK**

Click **Add** ↵

Lane Name (**Lane 2 right**)

Traffic Lane Properties

Eccentricity (**1.75**) ; Wheel Spacing (**1**) ; Lane Width(**3.5**)

Vehicular Load Distribution > **Lane Element**

Selection by > **2 Points ((0, 0, 0)(125, 0, 0))**

Click **OK**

When a traffic lane is curved or when the lane data entry with 2 Points becomes awkward due to discontinuity, select "Number" and directly type in the element numbers. (In this case, even if you select "Number" and input "1 to 53", the same traffic lanes are selected)

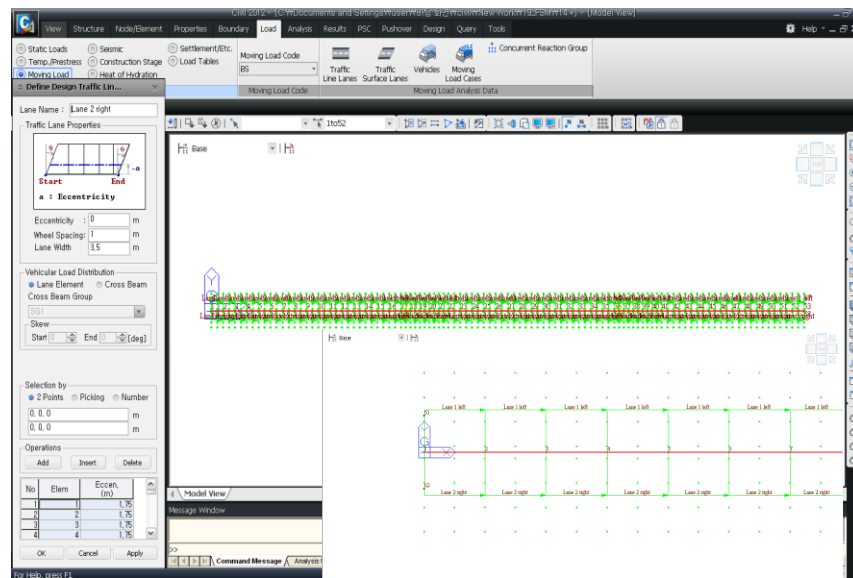


Figure 37. Traffic Lane Input Dialog & Input Result

► Definition of Vehicle Loads

Define the vehicles for live loads.

MIDAS/Civil contains the standard vehicle loads such as BS 5400, BS BD 37/01, AASHTO Standard, AASHTO LRFD, Caltrans, etc.

Load / Moving Load Analysis Data / Vehicles

Standard Name > **BD 37/01 Standard Load**
Vehicular Load Name > **HA & HB(Auto)**
Vehicular Load Type > **HA & HB(Auto)**

OK

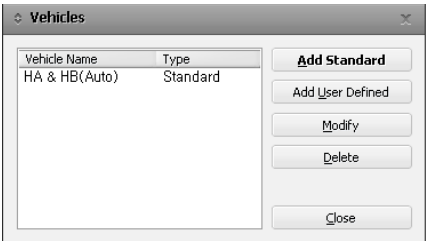


Figure 38. Definition of Vehicle Loads

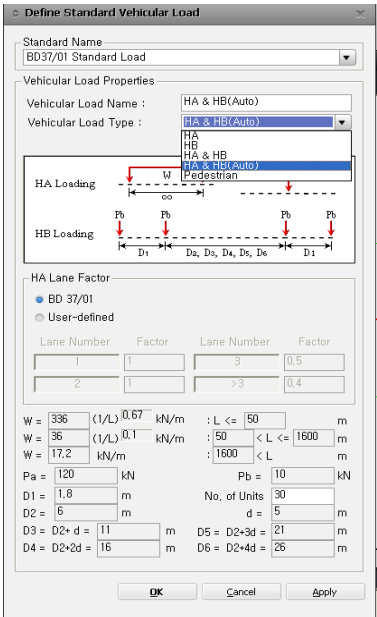


Figure 39. Definition of BD37/01 Standard Vehicular Load

► Conditions for applying live loads

To consider Load Cases, which combines the effects of HA and HB vehicle, Load case name MV U 1, MV U 2 3, MV S 1 and MV S 2 3 are created as below.

Load factors for HA loading for ULS, SLS, Combination 1 and Combinations 2 & 3 are taken from Section 6.2.7 of BD 37/01. Load factors for HB loading for ULS, SLS, Combination 1 and Combinations 2 & 3 are taken from Section 6.3.4 of BD 37/01. These load factors are automatically incorporated into moving load analysis results. Therefore, to avoid duplication, the user should not apply the load factors for moving loads while generating the Load Combinations.

Load Case Name		Type of Design Combination Factor	
		Ultimate Limit State	Serviceability Limit State
Combination of Loads	Combination 1	MV U 1	MV S 1
	Combination 2 & 3	MV U 2 3	MV S 2 3

Table 1. Definition of Load Case Name

Load / Moving Load Analysis Data /  **Moving Load Cases**

Click ↵

Load Case Name (**MV U 1**)

Check on **Auto Live Load Combination**

Type of Design Combination Factor>**Ultimate Limit State**

Combination of Loads>**Combination 1** ↵

Load Case Data

Scale Factor field (1) ; Number of Loaded Lanes (2)

Vehicle>**HA & HB (Auto)**

Assignment Lanes

List of Lanes (**Lane 1 left, Lane 2 right**) **Selected Lanes**

↵

↵

Load Case Name (**MV U 2 3**)

Type of Design Combination Factor>**Ultimate Limit State**

Combination of Loads>**Combination 2 or 3** ↵

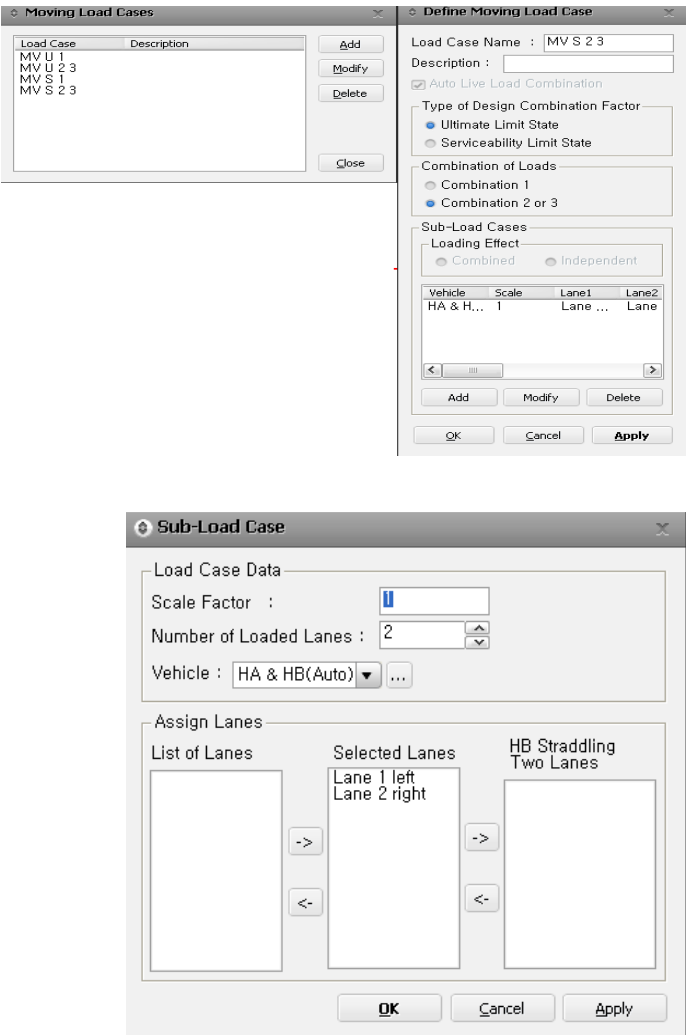











Figure 40. Definition of Live Load

Differential Settlement

► Definition of Differential Settlement Groups

Select the nodes, which can settle simultaneously, representing the abutments and piers, to individually define them as a Settlement Group.

Load / Settlement/Etc. /  **Settlement Group**
 Group Name > **A1** ; Settlement Displacement **(-0.01)**
 **Select By Window** (61, 62)  ↵
 Group Name > **P1** ; Settlement Displacement **(-0.01)**
 **Select By Window** (63, 64)  ↵
 Group Name > **P2** ; Settlement Displacement **(-0.01)**
 **Select By Window** (65, 66)  ↵
 Group Name > **A2** ; Settlement Displacement **(-0.01)**
 **Select By Window** (67, 68)  ↵

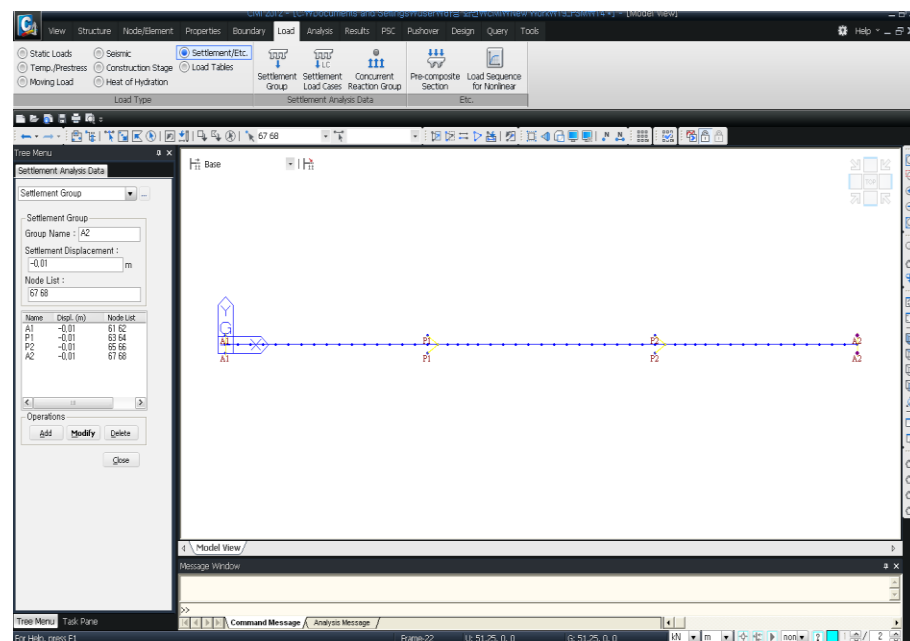





Figure 41. Definition of Differential Settlement Groups

► Conditions for Differential Settlement Loads

Using the data for differential settlement groups, the loading condition is defined.

Maximum/Minimum numbers of differential settlement groups are specified. Min: 1 support and Max: 3 supports are specified to investigate all the possible combinations of simultaneous settlements from which Min/Max results are produced.

Since the magnitude of the settlements of all 4 groups is identical, only a maximum of 3 combinations is used.

Load / Settlement/Etc. /  **Settlement Load Case**
 Load Case Name (SM)
 Select Settlement Group
 Settlement Group (A1, P1, P2, A2) -> **Selected Group**
 Smin (1) ; Smax (3)  

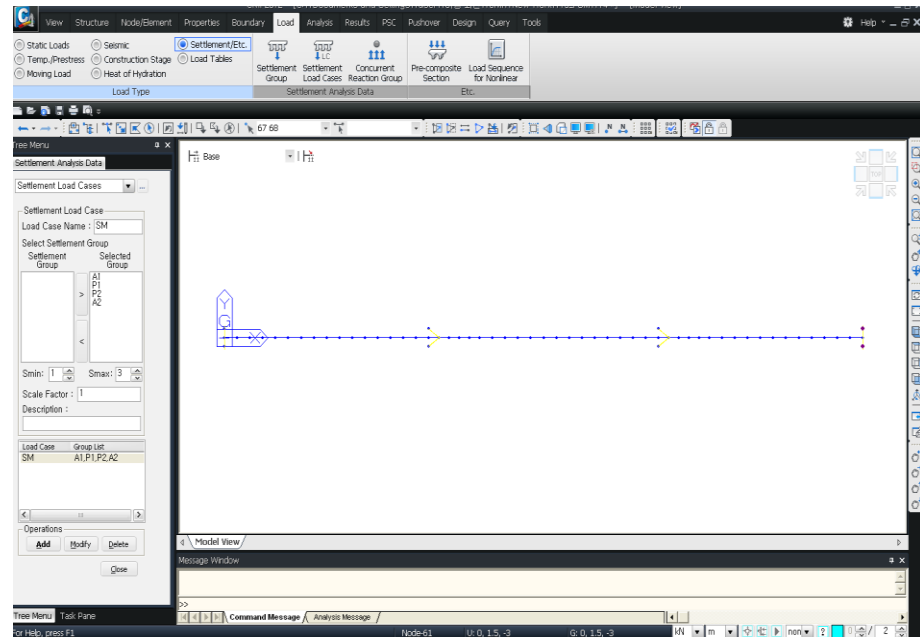



Figure 42. Definition of Loading Conditions for Differential Settlements

Definition of Construction Stages

We refer to the composition of construction stages outlined earlier to define the stages.


Concrete maturity (age) of 5 days is activated.

Load / Construction Stage /  **Define C.S**

 ↵

Name> **CS1**
Duration>**30**

Element tab

Group List>**SG1** ; Activation>Age (5) 


Boundary tab

Group List>**BG1**

Activation>Spring/Support Position>**Deformed (on)** 

Load tab

Group List>**Dead, PS1, Diaphragm1**

Activation>Active Day>**First**  ↵

Stage (days)		Element	Boundary	Load
CS1	30	SG1	BG1	Dead, PS1, Diaphragm1
CS2	30	SG2	BG2	PS2, Diaphragm2
CS3	30	SG3	BG3	PS3, Diaphragm3
CS4	10,000	-	-	Superimposed dead

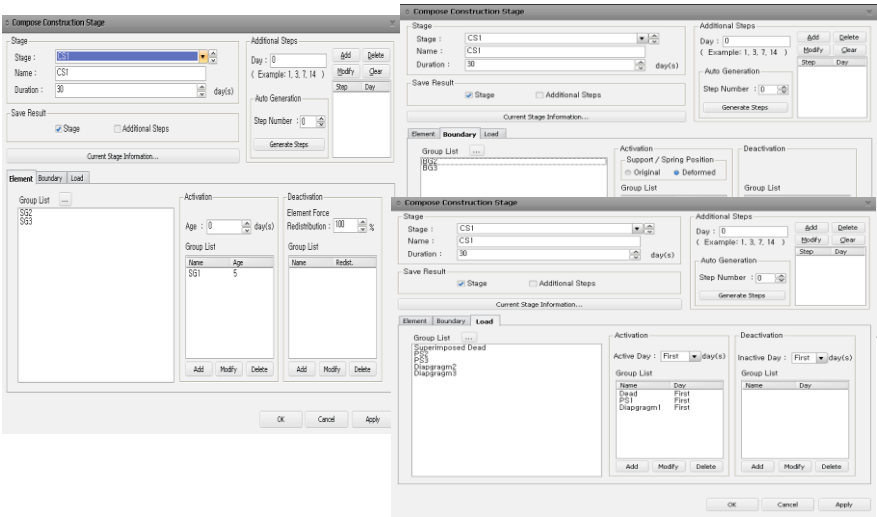


Figure 43. Dialog Boxes for defining Construction Stages

Performing Structural Analysis

Select the analysis options for construction stage analysis and moving load analysis and perform analysis.

Construction Stage Analysis

All the dead loads applied during the construction stages are included in CS:Dead Load. If results for other Load Cases need to be separated from CS:Dead Load, such Load Cases need to be selected in “Load Cases to be Distinguished from Dead Load for C.S. Output”. Separate results are then produced in CS:Erection Load.

Analysis / Construction Stage Analysis Control

Load Cases to be Distinguished from Dead Load for C.S. Output

Load Case>Superimposed Add

Save Output of Current Stage(Beam/Truss) (on) ⬇

Check on “Save Output of Current Stage (Beam /Truss)” to produce member forces generated only from each (current) stage. That is, not the member forces accumulated up to that (current) stage.

Checking on “Change with Tendon” in “Beam Section Property Change” will reflect the effect of tendons for calculating section properties by construction stages.

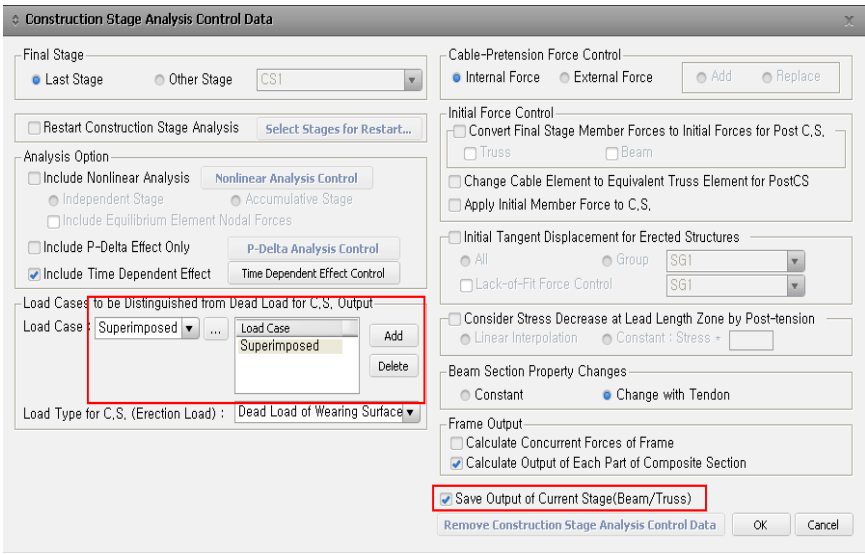


Figure 44. Construction Stage Analysis Control Data

Moving Load (Live Load) Analysis

Select the method of influence line calculation and the options for generation of analysis results.

Specify the number of points per beam element on which influence line is calculated. A number between 1 to 10 can be specified.

“Concurrent Force” will generate member forces, which take place simultaneously under the same loading.

Check on “Combined Stress” to generate combined stress results.

A substantial amount of results are generated from moving load analysis. Only the desired parts should be selected in groups for output generation.

Analysis / Moving Load Analysis Control

Influence Generating Method > Number/Line Element (2)

Analysis Results

Frame>Normal + Concurrent Force

Combined Stress Calculation (off)

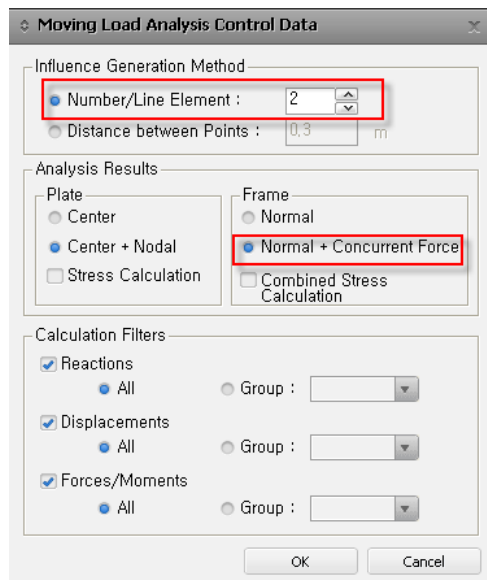


Figure 45. Moving Load Analysis Control Dialog

Execution of Structural Analysis

We have completed the process of structural modeling and defining the analysis options, so analysis can begin now.

Analysis / Perform Analysis

Checking Analysis Results

Construction stage analysis results will be reviewed via the versatile functionality of midas Civil.

Element Properties & Section Properties for each Construction Stage

The properties of each element used during the construction stages are produced in a table.

Select a stage to see the corresponding data; initial (Start) age, final (End) age, initial (Start) modulus of elasticity, final (End) modulus of elasticity, shrinkage accumulated up to the end of the corresponding stage and creep coefficient.

When a construction stage is selected, only the results pertaining to the corresponding stage are produced. The Post CS construction stage is selected followed by pressing the button to change the result values as below.

Results / Result Table / Construction Stage / Element Properties at Each Stage

PostCS (Post construction stage)

Stage	Start Age	End Age	Start Elasticity (N/m ²)	End Elasticity (N/m ²)	Cumulative Shrinkage	Creep Coeff.
1	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
2	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
3	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
4	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
5	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
6	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
7	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
8	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
9	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
10	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
11	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
12	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
13	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
14	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
15	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
16	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
17	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
18	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
19	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
20	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
21	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
22	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
23	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
24	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
25	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
26	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
27	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
28	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
29	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641
30	95.00	100.00	3750000.0000	3865769.3542	-0.0000	2.1641

Figure 46. Element Properties at each Construction Stage

Transformed section properties used in the last stage of the construction stage analysis are produced in a table. The properties may change with change in modulus of elasticity (if a time dependent material is used). And if tendons are included in sections, the tendon properties and the timing of grouting will affect the section properties.

- * In order to reflect the Tendon in section property calculations, "Change with Tendon" needs to be selected in Construction Stage Analysis Control.

Beam Section Property Changes

☐ Constant
 ☒ Change with Tendon

- * If "Change with Tendon" is selected, and "Bonded" type in "Tendon Property" is selected, the Tendon will be reflected in the section property calculations. Otherwise (in case of "Unbonded"), the Tendon is excluded and the net section is used in the calculations.

Bond Type

☒ Bonded
 ☐ Unbonded

The section properties at the last stage are used for calculating stresses due to additional loads applied at the completed stage such as moving load, temperature load, wind load, etc.

Results / Result Table / Construction Stage / Beam Section Properties at Last Stage

In the *.out file, we can see the section properties for all the stages in addition to those for the final stage.

Elem	Part	Area (m ²)	Ixx (m ⁴)	Iyy (m ⁴)	Izz (m ⁴)	Cyp (m)	Cym (m)	Czp (m)	Czm (m)	WArea (m ³)	Translational Distance (m)	
											Local-y	Local-z
1	I	6.2587	15.9796	7.8969	29.9433	4.2500	4.2500	1.2197	1.7863	6.2087	0.0000	-0.0043
1	J	6.2587	15.9796	7.9055	29.8836	4.2500	4.2500	1.2154	1.7846	6.2087	0.0000	-0.0060
2	I	6.2587	15.9796	7.9055	29.8836	4.2500	4.2500	1.2154	1.7846	6.2087	0.0000	-0.0060
2	J	6.2587	15.9796	7.9195	29.8239	4.2500	4.2500	1.2171	1.7829	6.2087	0.0000	-0.0077
3	I	6.2587	15.9796	7.9195	29.8239	4.2500	4.2500	1.2171	1.7829	6.2087	0.0000	-0.0077
3	J	6.2587	15.9796	7.9345	29.8164	4.2500	4.2500	1.2184	1.7816	6.2087	0.0000	-0.0090
4	I	6.2587	15.9796	7.9345	29.8164	4.2500	4.2500	1.2184	1.7816	6.2087	0.0000	-0.0090
4	J	6.2587	15.9796	7.9472	29.8111	4.2500	4.2500	1.2194	1.7806	6.2087	0.0000	-0.0099
5	I	6.2587	15.9796	7.9472	29.8111	4.2500	4.2500	1.2194	1.7806	6.2087	0.0000	-0.0099
5	J	6.2587	15.9796	7.9556	29.8080	4.2500	4.2500	1.2199	1.7801	6.2087	0.0000	-0.0105
6	I	6.2587	15.9796	7.9556	29.8080	4.2500	4.2500	1.2199	1.7801	6.2087	0.0000	-0.0105
6	J	6.2587	15.9796	7.9595	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
7	I	6.2587	15.9796	7.9595	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
7	J	6.2587	15.9796	7.9595	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
8	I	6.2587	15.9796	7.9595	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
8	J	6.2587	15.9796	7.9595	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
9	I	6.2587	15.9796	7.9595	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
9	J	6.2587	15.9796	7.9594	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
10	I	6.2587	15.9796	7.9594	29.8070	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
10	J	6.2587	15.9796	7.9597	29.8069	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
11	I	6.2587	15.9796	7.9597	29.8069	4.2500	4.2500	1.2201	1.7799	6.2087	0.0000	-0.0107
11	J	6.2587	15.9796	7.9488	29.8102	4.2500	4.2500	1.2195	1.7805	6.2087	0.0000	-0.0101
12	I	6.2587	15.9796	7.9488	29.8102	4.2500	4.2500	1.2195	1.7805	6.2087	0.0000	-0.0101
12	J	6.2587	15.9796	7.9225	29.8203	4.2500	4.2500	1.2177	1.7823	6.2087	0.0000	-0.0083
13	I	6.2587	15.9796	7.9225	29.8203	4.2500	4.2500	1.2177	1.7823	6.2087	0.0000	-0.0083
13	J	6.2587	15.9796	7.8904	29.8376	4.2500	4.2500	1.2146	1.7854	6.2087	0.0000	-0.0052
14	I	6.2587	15.9796	7.8904	29.8376	4.2500	4.2500	1.2146	1.7854	6.2087	0.0000	-0.0052
14	J	6.2587	15.9796	7.8698	29.8638	4.2500	4.2500	1.2102	1.7898	6.2087	0.0000	-0.0008
15	I	6.2587	15.9796	7.8698	29.8638	4.2500	4.2500	1.2102	1.7898	6.2087	0.0000	-0.0008
15	J	6.2587	15.9796	7.8765	29.8696	4.2500	4.2500	1.2060	1.7940	6.2087	0.0000	0.0034
16	I	6.2587	15.9796	7.8765	29.8696	4.2500	4.2500	1.2060	1.7940	6.2087	0.0000	0.0034
16	J	6.2587	15.9796	7.8827	29.8867	4.2500	4.2500	1.2045	1.7955	6.2087	0.0000	0.0048

Figure 47. Section Property Data at the Last Stage

Checking Construction Stage Member Forces & Stresses

Member forces can be checked in a diagram using the Beam Diagram function. If a beam element is selected after invoking Quick View, member forces at any particular point on the selected element can be checked in detail.

Results / Forces /  Beam Diagrams...

CS4

Load Cases/Combinations>**CS:Summation** ; Step>**Last Step**

Components>**My**

Display Options> **Solid Fill**

Type of Display

Contour (on) ; **Legend (on)**  ↵

Type of Display>**Quick View**

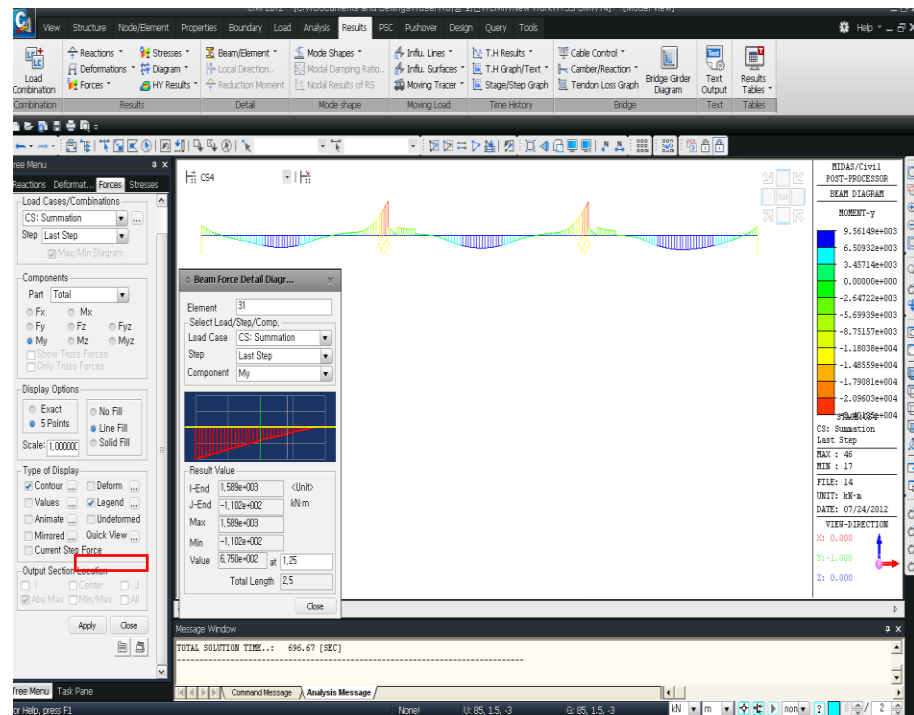


Figure 48. Checking Member Forces at CS4

Using the Beam Stresses(PSC) function, the stresses in a PSC section can be checked in a diagram. A total of 10 locations, Top/Bot vetices (1 to 4), Center (7 & 8) and shear checking points (5, 6, 9 & 10) defined at the time of defining the PSC section, can be checked.

Let us check the bottom chord stress for CS:Summation at the last construction stage.

Results / Stresses /  Beam Stresses(PSC)...

CS4


Load Cases/Combinations>**CS:Summation** ; Step>**Last Step**

Section Position>**Position 3**

Components>**Sig-xx(Summation)**

Type of Display

Contour (on) ; Legend (on)

 **Top-Bot chord stresses for each construction stage can be also checked in Bridge Girder Diagram. In case of a PSC section, Beam Stresses (PSC) can be used to check Contour in the Model View state.**

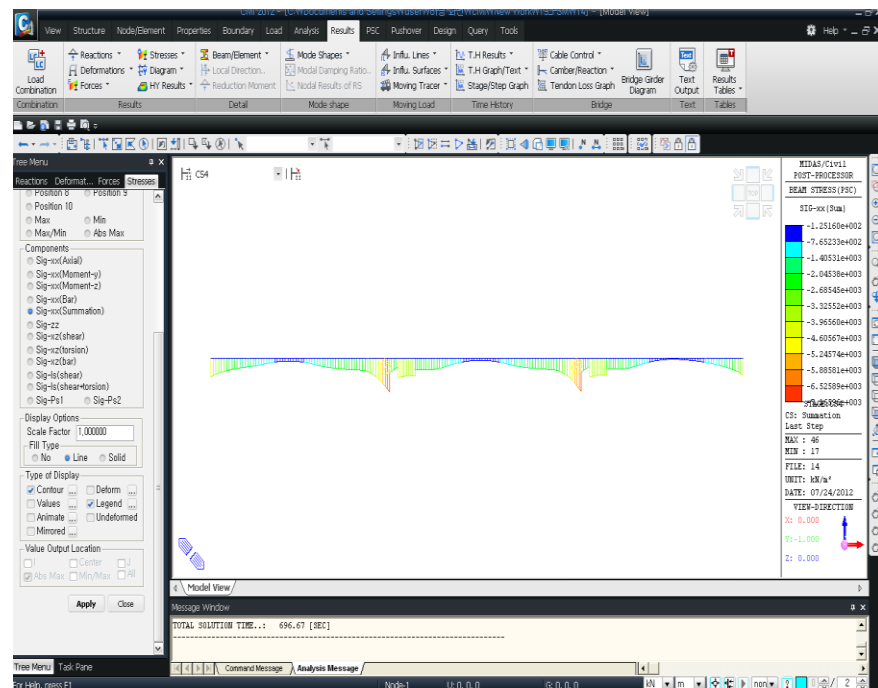


Figure 49. Bottom Chord Stresses at the Last Stage

Using User Defined Diagram, different results (displacements / member forces / stresses) for different elements/groups can be produced.

We will generate results for displacements in the left span, bending moments in the middle span and stresses in the right span in a single diagram simultaneously. Let us check displacements / member forces / stresses for CS:Summation at the last construction stage.

Combined results can be produced only in the same construction stage.

Output option can be selected in Modify Display Option

Results / Diagram / Define Diagram...

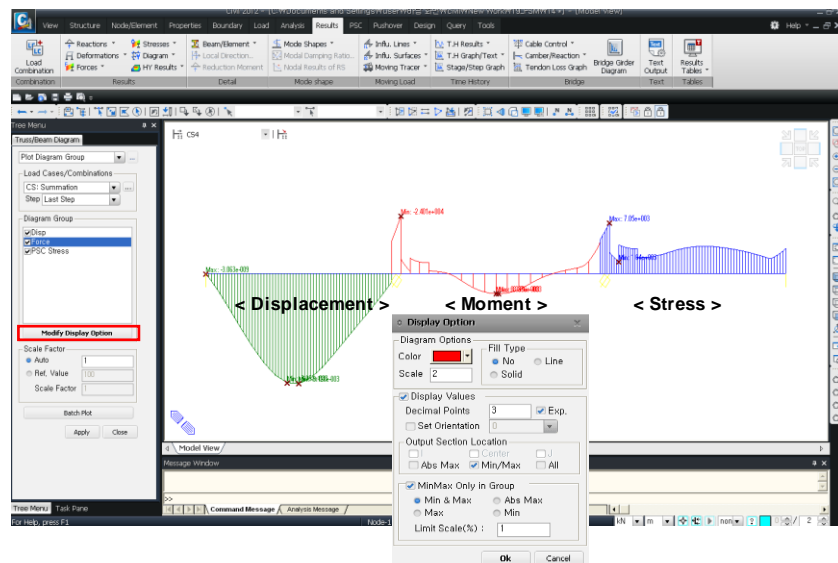
CS4

Element>1to16 ; Type of Result>Displacement
 Component>DZ ; Group Name>Disp Add
 Element>17to35 ; Type of Result>Beam Force/Moments
 Component>My ; Group Name>Force Add
 Element>36to52 ; Type of Result>Beam Stresses(PSC)
 Section Position>Abs Max ; Components>Sig-xx(Summation)
 Group Name>PSC Stress Add

Results / Diagram / Plot Diagram...

Load Cases/Combination>CS:Summation

Diagram Group>Disp(on), Force(on), PSC Stress(on)



Note that the three plots above have different scale factors to properly display in this figure. In order to check the results, you may enlarge the figure and compare the values.

Figure 50. User Defined Diagram Output Display

Checking Results using Graphs

The change in stresses with the progress of construction stages in the support element (No.36) will be checked in a Graph.

Results / Stage/Step History Graph

CS4

Define Function>**Beam Force/Stress** Add New Function ↵

Name(**36_ax**) ; Element No.(**36**) ; **Stress**

Point>**I-Node** ; Components>**Axial** ↵

Name(**36_b(+y)**) ; Components>**Bend(+y)** ↵

Name(**36_b(-y)**) ; Components>**Bend(-y)** ↵

Name(**36_b(+z)**) ; Components>**Bend(+z)** ↵

Name(**36_b(-z)**) ; Components>**Bend(-z)** ↵

Mode >**Multi Func.** 🗣

Step Option>**All Steps**

Load Cases/Combinations>**Summation** ↵

🗣 If “Multi LCase” is selected, the results history of the component of the corresponding element for a number of Load Cases can be checked.

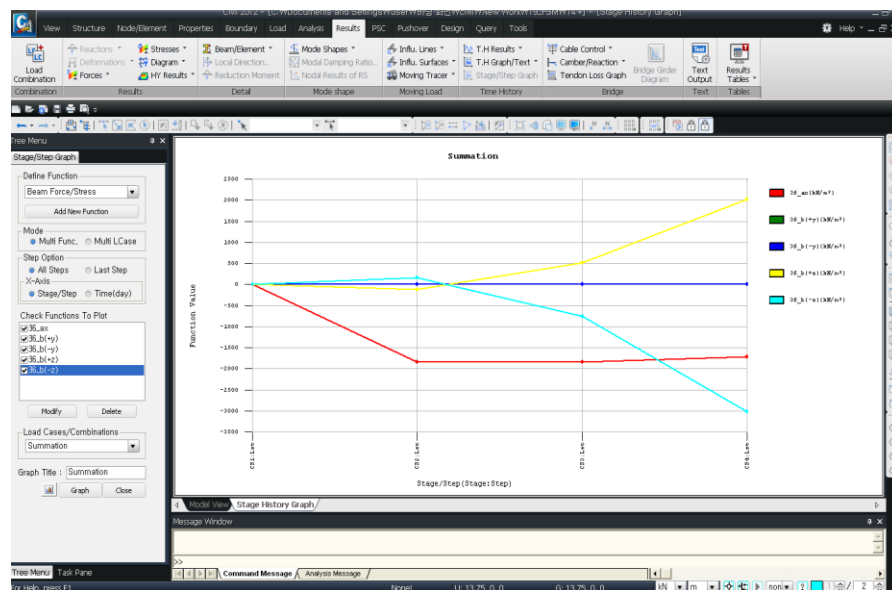


Figure 51. Change in Stresses with Construction Stages

Checking Results using Tables

Tables are also useful in checking construction stage analysis results. Tables can be manipulated in various ways by right-clicking on the tables.

From “Records Activation Dialog”, tables can be generated by selecting elements to be checked for stresses, load cases, construction stages (steps), elements on which points of stress output are required, load cases, construction stages (steps), stress output locations on elements, stress output locations on a section, etc.

The Sorting Dialog allows us to sort/arrange the data based on the sorting criteria. The Style Dialog allows us to change the data type and produce results.

Let us check top vertex stresses for CS:Summation at the last construction stage.

Results / Result Tables / Beam / Stresses(PSC)

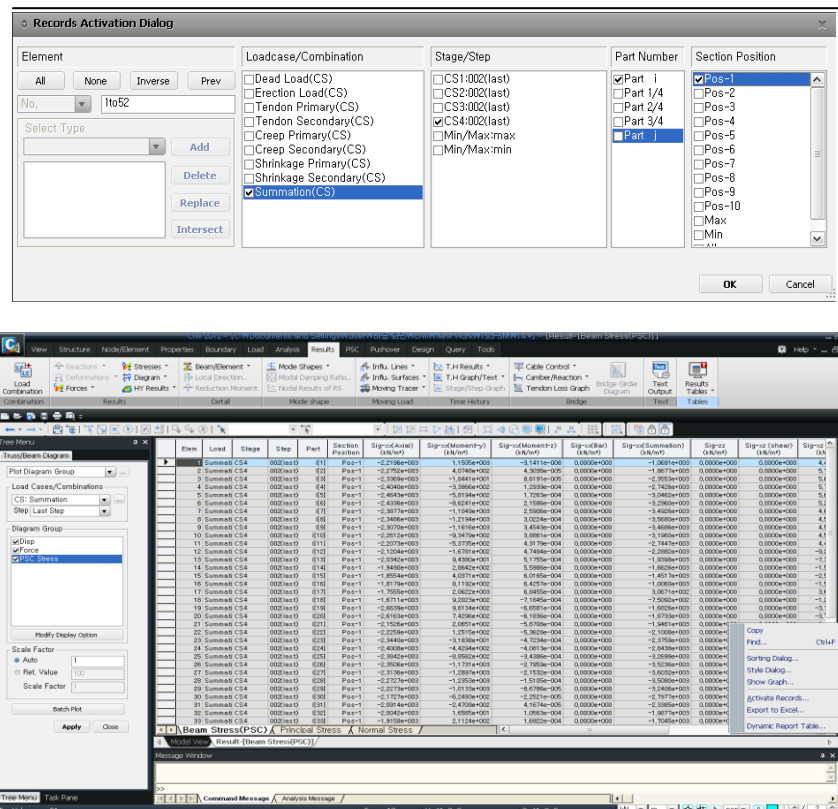
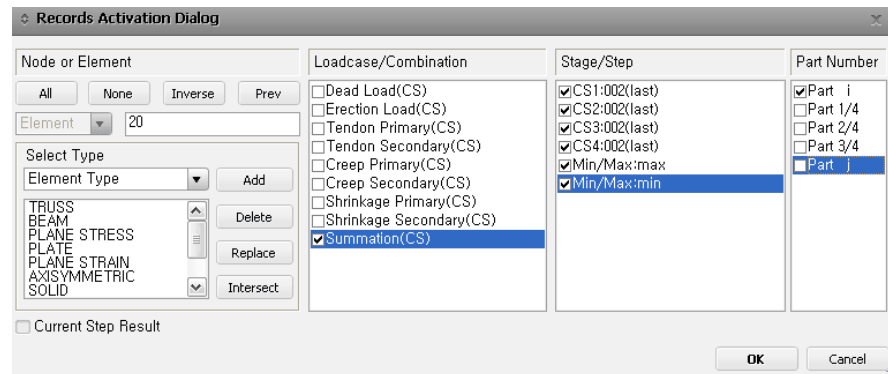


Figure 52. Checking Top Chord Stresses using Table

In “Construction Stage Analysis Control” dialog box, if “**Save Output of Current Stage (Beam/Truss)**” ☒ Save Output of Current Stage(Beam/Truss) option has been checked on, we can generate the member forces resulting only from the corresponding construction stage (not the member forces accumulated up to that stage). So in order to produce results for the un-accumulated effects of one given construction stage, check “Current Step Result” for all the stages.

Results / Result Tables / Beam /  Force



Records Activation Dialog

Node or Element: All None Inverse Prev
Element: 20

Select Type: Element Type Add Delete Replace Intersect

Loadcase/Combination: ☐ Dead Load(CS) ☐ Erection Load(CS) ☐ Tendon Primary(CS) ☐ Tendon Secondary(CS) ☐ Creep Primary(CS) ☐ Creep Secondary(CS) ☐ Shrinkage Primary(CS) ☐ Shrinkage Secondary(CS) ☒ Summation(CS)

Stage/Step: ☒ CS1:002(last) ☒ CS2:002(last) ☒ CS3:002(last) ☒ CS4:002(last) ☒ Min/Max:max ☒ Min/Max:min

Part Number: ☒ Part i ☐ Part 1/4 ☐ Part 2/4 ☐ Part 3/4 ☒ Part j

☐ Current Step Result

OK Cancel

Elem	Load	Stage	Step	Part	Axial (kN)	Shear-y (kN)	Shear-z (kN)	Torsion (kN-m)	Moment-y (kN-m)	Moment-z (kN-m)
20	Summati CS1	002(last)	I[20]		-10296,16	0,00	1802,79	0,00	-1447,41	0,00
20	Summati CS2	002(last)	I[20]		-17587,18	-0,00	253,98	0,00	-1650,80	-0,00
20	Summati CS3	002(last)	I[20]		-17507,52	-0,00	319,25	0,00	-1514,36	-0,01
20	Summati CS4	002(last)	I[20]		-16096,67	-0,00	-485,96	0,00	-4833,07	-0,00
20	Summati Min/Max	max	I[20]		-10296,16	0,00	1848,67	0,00	-1083,72	0,00
20	Summati Min/Max	min	I[20]		-17786,08					-0,01

Accumulated member forces at each construction stage

Elem	Load	Stage	Step	Part	Axial (kN)	Shear-y (kN)	Shear-z (kN)	Torsion (kN-m)	Moment-y (kN-m)	Moment-z (kN-m)
20	Summati CS1	002(last)	I[20]		97,94	0,00	-21,12	0,00	10,11	0,00
20	Summati CS2	002(last)	I[20]		68,91	-0,00	-15,28	0,00	-104,60	-0,00
20	Summati CS3	002(last)	I[20]		41,75	-0,00	-4,99	0,00	-92,17	-0,00
20	Summati CS4	002(last)	I[20]		125,97	0,00	-21,98	-0,00	-43,41	0,00
20	Summati Min/Max	max	I[20]		0,00	0,00	0,00	-0,00	0,00	0,00
20	Summati Min/Max	min	I[20]		0,00					0,00

Member forces solely due to the corresponding stage

Figure 53. Member Forces due to the sole effect of Current Stage (below)

Prestress Losses

We can check the change in tendon tension at each construction stage due to prestress losses.

In the “Tendon Time Dependent Loss Graph” dialog box, only the tendons included in the stage selected in the “Stage” selection window can be checked. A Graph is generated for selected tendons, selected Stage and selected Step. Click

Animate to check the results in an animation.

Results /Tendon Time-dependent Loss Graph

Tendon>**A1L** **Animate** ↵

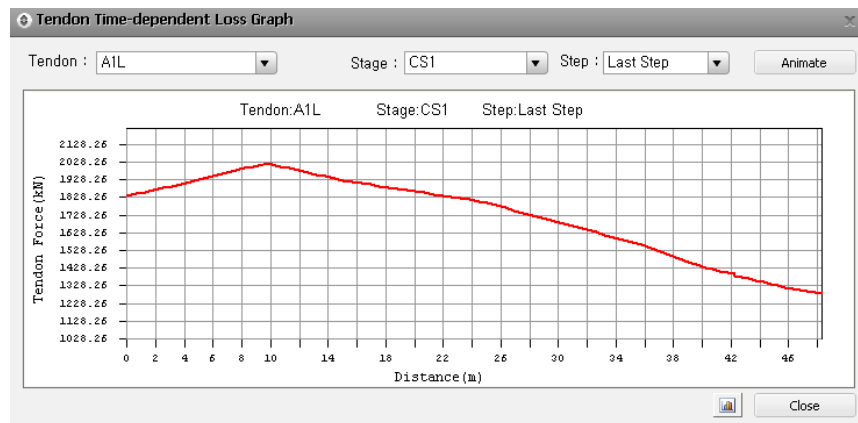


Figure 54. Graph showing Loss of Prestress Forces

Checking Tendon Information

The tendon information used in construction stage analysis can be produced in a table.

The coordinates of the tendons placed in elements are produced.

Results / Result Tables / Tendon / Tendon Coordinates...

	Tendon Name	No	x (m)	y (m)	z (m)
►	A1L	0	0,0000	0,0000	0,0000
	A1L	1	0,0000	2,4660	-1,0000
	A1L	2	0,6250	2,4502	-1,0790
	A1L	3	1,2500	2,4341	-1,1593
	A1L	4	1,8750	2,4176	-1,2420
	A1L	5	2,5000	2,4004	-1,3281
	A1L	6	3,1250	2,3829	-1,4157
	A1L	7	3,7500	2,3656	-1,5018
	A1L	8	4,3750	2,3493	-1,5837
	A1L	9	5,0000	2,3340	-1,6602
	A1L	10	5,6250	2,3197	-1,7316
	A1L	11	6,2500	2,3063	-1,7984
	A1L	12	6,8750	2,2939	-1,8607

Figure 55. Tendon Coordinates Table

Elongation of tendons is produced. Timing of tensioning each tendon, elongation of tendons and elements at the start and end points of the tendons and their sum are produced.

Results / Result Tables / Tendon / Tendon Elongations...

	Tendon Name	Stage	Step	Tendon Elongation		Element Elongation		Summation	
				Begin (m)	End (m)	Begin (m)	End (m)	Begin (m)	End (m)
►	A1L	CS1	001(first	0,2766	0,0000	0,0005	0,0000	0,2770	0,0000
	A1R	CS1	001(first	0,2766	0,0000	0,0005	0,0000	0,2770	0,0000
	A2L	CS1	001(first	0,2785	0,0000	0,0005	0,0000	0,2789	0,0000
	A2R	CS1	001(first	0,2785	0,0000	0,0005	0,0000	0,2789	0,0000
	A3L	CS1	001(first	0,2799	0,0000	0,0005	0,0000	0,2803	0,0000
	A3R	CS1	001(first	0,2799	0,0000	0,0005	0,0000	0,2803	0,0000
	A4L	CS1	001(first	0,2825	0,0000	0,0005	0,0000	0,2829	0,0000
	A4R	CS1	001(first	0,2825	0,0000	0,0005	0,0000	0,2829	0,0000
	B1L	CS2	001(first	0,2569	0,0000	0,0004	0,0000	0,2573	0,0000
	B1R	CS2	001(first	0,2569	0,0000	0,0004	0,0000	0,2573	0,0000
	B2L	CS2	001(first	0,2509	0,0000	0,0004	0,0000	0,2513	0,0000
	B2R	CS2	001(first	0,2509	0,0000	0,0004	0,0000	0,2513	0,0000
	B3L	CS2	001(first	0,2818	0,0000	0,0005	0,0000	0,2822	0,0000
	B3R	CS2	001(first	0,2818	0,0000	0,0005	0,0000	0,2822	0,0000
	B4L	CS2	001(first	0,2848	0,0000	0,0005	0,0000	0,2853	0,0000

Figure 56. Tendon Elongation Table

The effective stresses and effective prestressing force in the tendons can be checked by group and construction stage. Vertical and horizontal force components of the tendons can be readily obtained from the distance from the centroid of the section to the tendon group and the orientation of the tendon (direction cosine).

Results / Result Tables / Tendon Arrangement...

Select a construction stage and click **Apply** to produce the results corresponding to the stage.

Elem	Part	Tendon Number	Yp (m)	Zp (m)	Average Sin θ ([deg])	Average Cos θ ([deg])	Average Stress (kN/m ²)	Average Force (kN)
The arrangement data for tendon group [A1] at the stage of [CS1]								
Tendon Group		A1	Stage		CS1	Apply		
1	I	2	0.0000	0.2094	-0.1301	-0.9915	1140860,7379	1838,1548
1	J	2	0.0000	-0.1187	-0.1301	-0.9915	1166973,4564	1880,2276
2	I	2	0.0000	-0.1187	-0.1317	-0.9913	1166973,7832	1880,2282
2	J	2	0.0000	-0.4507	-0.1317	-0.9913	1196465,8749	1927,7458
3	I	2	0.0000	-0.4507	-0.1028	-0.9947	1196466,3047	1927,7465
3	J	2	0.0000	-0.7090	-0.1028	-0.9947	1226038,9033	1978,6163
4	I	2	0.0000	-0.7090	-0.0736	-0.9973	1226038,2614	1978,6169
4	J	2	0.0000	-0.8934	-0.0736	-0.9973	1250906,0441	2015,4598
5	I	2	0.0000	-0.8934	-0.0442	-0.9990	1250906,3049	2015,4602
5	J	2	0.0000	-1.0039	-0.0442	-0.9990	1222161,0060	1969,1458
6	I	2	0.0000	-1.0039	-0.0147	-0.9999	1222161,1722	1969,1461
6	J	2	0.0000	-1.0406	-0.0147	-0.9999	1194626,3057	1924,7819
7	I	2	0.0000	-1.0406	0.0000	1.0000	1194626,3532	1924,7820
7	J	2	0.0000	-1.0405	0.0000	1.0000	1174713,4298	1892,6983
8	I	2	0.0000	-1.0405	-0.0000	-1.0000	1174713,4298	1892,6983
8	J	2	0.0000	-1.0406	-0.0000	-1.0000	1155564,6599	1861,8458
9	I	2	0.0000	-1.0406	0.0001	1.0000	1155564,6595	1861,8458
9	J	2	0.0000	-1.0404	0.0001	1.0000	1136158,1002	1830,5779

Figure 57. Tendon Arrangement Table

The effective stresses & forces in the table above are the results reflecting both immediate and long-term losses of the tendon. If the effective prestress forces for the immediate losses (friction, anchorage slip & elastic shortening) other than the long-term losses are of interest, right-click on the table and check the forces from "Tendon Immediate Loss Graph".

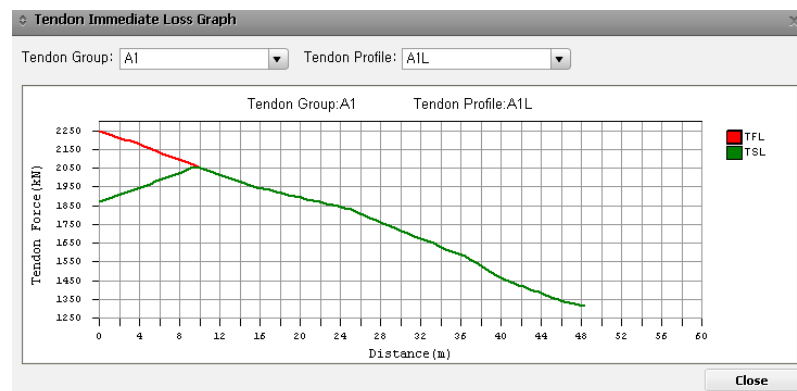


Figure 58. Tendon Force due to Immediate Loss

For each tendon group, losses to due friction, anchorage slip, elastic shortening, creep, shrinkage, relaxation, etc. are separately classified in a table.

Results Tab / Result Tables / Tendon Loss...

Select a construction stage and click **Apply** to produce the results corresponding to the stage.

Elem	Part	Stress (After Immediate Loss) : A (kN/m²)	Elastic Deform. Loss : B (kN/m²)	Stress(Elasti c Loss)/ Stress(Immed iate Loss)	Creep/Shrinkage Loss (kN/m²)	Relaxation Loss (kN/m²)	Stress(After All Loss)/ Stress(After Immediate Loss)	Effective Num.
The Loss of tendon group [A1] at the stage of [CS1]								
Tendon Group B3		Stage CS4		CS2 CS3 CS4	Apply			
1 I	1161959.6511	267.1150			-13576.4332	-7789.5949	0.9818	2.0000
1 J	1189636.2538	315.7131			-15003.3760	-7975.1345	0.9809	2.0000
2 I	1189636.2538	315.7089			-15003.0450	-7975.1345	0.9810	2.0000
2 J	1219836.6206	396.7800		1.0003	-15589.9327	-8177.5930	0.9808	2.0000
3 I	1219836.6206	396.7729		1.0003	-15589.4958	-8177.5930	0.9808	2.0000
3 J	1251591.3959	479.3410		1.0004	-15841.3615	-8390.4721	0.9812	2.0000
4 I	1251591.3959	479.3338		1.0004	-15840.8962	-8390.4721	0.9812	2.0000
4 J	1274265.1849	544.6933		1.0004	-15361.3604	-8542.4736	0.9817	2.0000
5 I	1274265.1849	544.6872		1.0004	-15361.0936	-8542.4736	0.9817	2.0000
5 J	1244113.8475	552.5986		1.0004	-14165.0963	-8340.3438	0.9824	2.0000
6 I	1244113.8475	552.5943		1.0004	-14164.9259	-8340.3438	0.9824	2.0000
6 J	1215069.1930	521.4844		1.0004	-12818.7387	-8145.6330	0.9832	2.0000
7 I	1215069.1930	521.4831		1.0004	-12818.6900	-8145.6330	0.9832	2.0000
7 J	1194131.8609	488.5435		1.0004	-11901.7021	-8005.2724	0.9837	2.0000
8 I	1194131.8609	488.5435		1.0004	-11901.7022	-8005.2724	0.9837	2.0000
8 J	1174479.8756	473.5575		1.0004	-11515.2447	-7873.5285	0.9839	2.0000
9 I	1174479.8756	473.5575		1.0004	-11515.2451	-7873.5285	0.9839	2.0000
9 J	1155066.6314	475.5742		1.0004	-11640.7201	-7743.3852	0.9836	2.0000
10 I	1155066.6314	475.5740		1.0004	-11640.7152	-7743.3852	0.9836	2.0000
10 J	1134147.6817	494.6497		1.0004	-12265.3886	-7603.1479	0.9829	2.0000
11 I	1134147.6817	494.6557		1.0004	-12265.6237	-7603.1479	0.9829	2.0000
11 J	1099333.9226	465.1451		1.0004	-12557.5779	-7369.7619	0.9823	2.0000
12 I	1099333.9226	465.1616		1.0004	-12558.2697	-7369.7619	0.9823	2.0000
12 J	1064511.0072	367.9269		1.0003	-12444.9432	-7136.3145	0.9820	2.0000
13 I	1064511.0072	367.9462		1.0003	-12445.9457	-7136.3145	0.9820	2.0000
13 J	1031565.9112	262.1179		1.0003	-12472.6840	-6915.4558	0.9815	2.0000
14 I	1031565.9112	262.1380		1.0003	-12474.1168	-6915.4558	0.9815	2.0000
14 J	996879.1124	220.5014		1.0002	-13507.6980	-6682.9209	0.9800	2.0000
15 I	996879.1124	220.5053		1.0002	-13508.0415	-6682.9209	0.9800	2.0000

Figure 59. Tendon (Tension) Loss Table

Right-click on the table and select “Tendon Time-dependent Loss Graph” to check the effective prestress forces after accounting for tension losses.

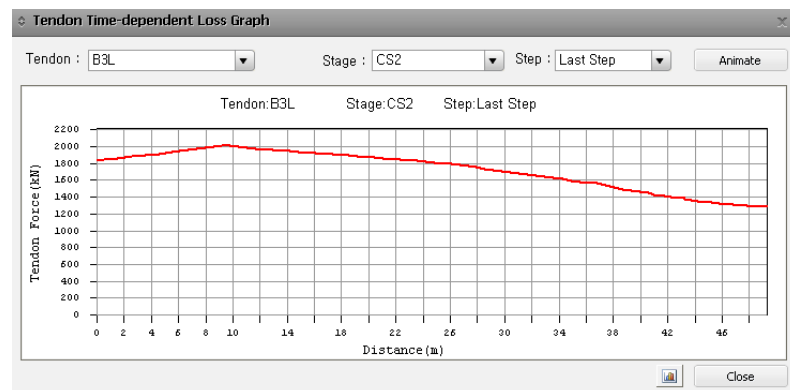



Figure 60. Tendon Time-dependent Loss Graph

Tendon type, property and weight for each group can be tabulated.

Results / Result Tables / Tendon / Tendon Weight...

PostCS

 Tendon Weight
can be produced
only in the PostCS
stage.

	Tendon Name	Tendon Num	Area (m²)	Length (m)	Weight/Length (kN/m)	Weight (kN)	Total Weight (kN)
▶	A1L	1,00	0,001611	48,320358	0,126479	6,111520	6,111520
	A1R	1,00	0,001611	48,320358	0,126479	6,111520	6,111520
	A2L	1,00	0,001611	48,345183	0,126479	6,114660	6,114660
	A2R	1,00	0,001611	48,345183	0,126479	6,114660	6,114660
	A3L	1,00	0,001611	48,282773	0,126479	6,106767	6,106767
	A3R	1,00	0,001611	48,282773	0,126479	6,106767	6,106767
	A4L	1,00	0,001611	48,323420	0,126479	6,111908	6,111908
	A4R	1,00	0,001611	48,323420	0,126479	6,111908	6,111908
	B1L	1,00	0,001611	45,341275	0,126479	5,734728	5,734728
	B1R	1,00	0,001611	45,341275	0,126479	5,734728	5,734728
	B2L	1,00	0,001611	45,381107	0,126479	5,739766	5,739766
	B2R	1,00	0,001611	45,381107	0,126479	5,739766	5,739766
	B3L	1,00	0,001611	49,312943	0,126479	6,237062	6,237062
	B3R	1,00	0,001611	49,312943	0,126479	6,237062	6,237062
	B4L	1,00	0,001611	49,339207	0,126479	6,240383	6,240383
	B4R	1,00	0,001611	49,339207	0,126479	6,240383	6,240383
	C1L	1,00	0,001611	32,159596	0,126479	4,067520	4,067520
	C1R	1,00	0,001611	32,159596	0,126479	4,067520	4,067520
	C2L	1,00	0,001611	32,115616	0,126479	4,061957	4,061957
	C2R	1,00	0,001611	32,115616	0,126479	4,061957	4,061957
	C3L	1,00	0,001611	36,068159	0,126479	4,561872	4,561872
	C3R	1,00	0,001611	36,068159	0,126479	4,561872	4,561872
	C4L	1,00	0,001611	36,026608	0,126479	4,556617	4,556617
	C4R	1,00	0,001611	36,026608	0,126479	4,556617	4,556617
	SUM	24,00	-	1038,032491	0,126479	-	131,289519

Figure 61. Tendon Weight Table

Checking Moving Load Analysis Results

The member forces produced in moving load analysis are the results of maximum values for each component in the corresponding element. As such, the locations of the loads causing each maximum force component may be different.

In order to obtain the concurrent member forces, right-click on the table and use the "View by Max Value Item" function. We can then check the corresponding force components associated with one maximum force component.

When Moment-y is maximum, other force components occurring at the same time are produced.

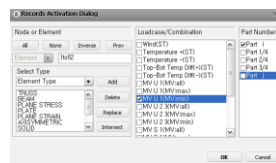
Results / Result Tables / Beam /  Force

Loadcase/Combination>MV U 1(MV:min) ; Part Number>Part I ↵

(Context Menu) View by Max Value Item

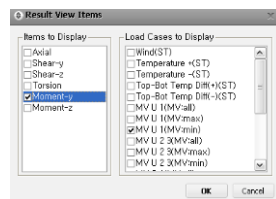
Items to Display>Moment-y

Load Cases to Display> MV U 1(MV:min) ↵



Elem	Load	Part	Axial (kN)	Shear-y (kN)	Shear-z (kN)	Torsion (kN m)	Moment-y (kN m)	Moment-z (kN m)
1	MV U 1	I(1)	-0.16	-0.09	-2152.40	-2442.67	-0.00	-1.49
2	MV U 1	I(2)	-0.17	-0.09	-1960.93	-2231.11	-603.84	-1.36
3	MV U 1	I(3)	-0.15	-0.09	-1756.05	-2200.67	-1207.08	-1.33
4	MV U 1	I(4)	-0.13	-0.09	-1567.91	-2104.52	-1810.61	-1.20
5	MV U 1	I(5)	-0.11	-0.09	-1386.10	-2016.92	-2414.15	-1.18
6	MV U 1	I(6)	-0.05	-0.09	-1211.33	-1934.47	-3017.69	-1.23
7	MV U 1	I(7)	-0.00	-0.09	-1044.29	-1856.44	-3621.23	-1.37
8	MV U 1	I(8)	-0.00	-0.09	-865.66	-1782.33	-4224.77	-1.60
9	MV U 1	I(9)	-0.00	-0.09	-736.14	-1711.74	-4828.30	-1.82
10	MV U 1	I(10)	-0.01	-0.09	-597.49	-1648.06	-5431.84	-2.05
11	MV U 1	I(11)	-0.09	-0.09	-468.90	-1586.41	-6035.36	-2.32
12	MV U 1	I(12)	-0.18	-0.09	-351.24	-1680.35	-6638.82	-3.66
13	MV U 1	I(13)	-0.19	-0.09	-245.05	-1783.08	-7242.46	-4.65
14	MV U 1	I(14)	-0.15	-0.09	-150.98	-1895.26	-7850.23	-6.14
15	MV U 1	I(15)	-0.11	-0.09	-81.48	-2014.69	-8460.70	-8.10
16	MV U 1	I(16)	-0.03	-0.09	-66.25	-2145.80	-11084.62	-4.20
17	MV U 1	I(17)	-0.05	-0.09	-60.23	-2282.01	-14005.03	-3.55
18	MV U 1	I(18)	-0.26	-0.19	-2287.31	-2477.43	-11707.35	-8.40
19	MV U 1	I(19)	-0.54	-0.19	-2148.72	-2595.54	-9753.36	-10.03
20	MV U 1	I(20)	-0.61	-0.19	-2003.18	-2243.36	-8316.63	-10.07
21	MV U 1	I(21)	-0.96	-0.19	-1906.15	-2137.61	-7541.56	-7.79
22	MV U 1	I(22)	-0.25	-0.19	-1781.76	-2043.50	-7024.62	-6.43
23	MV U 1	I(23)	-0.07	-0.19	-1624.04	-1933.28	-6504.59	-4.87
24	MV U 1	I(24)	-0.00	-0.19	-1464.94	-1832.42	-6103.68	-4.24
25	MV U 1	I(25)	-0.00	-0.19	-1305.78	-1736.67	-5830.90	-3.98
26	MV U 1	I(26)	-0.00	-0.00	-0	-0	-0	-0
27	MV U 1	I(27)	-0.00	-0.00	-0	-0	-0	-0
28	MV U 1	I(28)	-0.00	-0.00	-0	-0	-0	-0
29	MV U 1	I(29)	-0.00	-0.00	-0	-0	-0	-0
30	MV U 1	I(30)	-0.11	-0.19	-600.00	-1860.10	-5126.12	-2.12

Max member force



Elem	Load	Part	Component	Axial (kN)	Shear-y (kN)	Shear-z (kN)	Torsion (kN m)	Moment-y (kN m)	Moment-z (kN m)
1	MV U 1	I(1)	Moment-y	1.03	0.01	-1607.54	-255.72	-1.00	-0.22
2	MV U 1	I(2)	Moment-y	-0.16	0.00	241.42	2.30	-603.84	-0.00
3	MV U 1	I(3)	Moment-y	-0.13	0.00	241.42	2.30	-1207.08	-0.00
4	MV U 1	I(4)	Moment-y	-0.09	0.00	241.42	2.30	-1810.61	-0.00
5	MV U 1	I(5)	Moment-y	-0.06	0.00	241.42	2.30	-2414.15	-0.01
6	MV U 1	I(6)	Moment-y	-0.02	0.00	241.42	2.30	-3017.69	-0.01
7	MV U 1	I(7)	Moment-y	0.00	0.00	241.42	2.30	-3621.23	-0.01
8	MV U 1	I(8)	Moment-y	0.00	0.00	241.42	2.30	-4224.77	-0.01
9	MV U 1	I(9)	Moment-y	0.00	0.00	241.42	2.30	-4828.30	-0.01
10	MV U 1	I(10)	Moment-y	-0.00	0.00	241.42	2.30	-5431.84	-0.01
11	MV U 1	I(11)	Moment-y	0.06	0.00	241.42	2.30	-6035.36	-0.02
12	MV U 1	I(12)	Moment-y	0.18	0.00	241.42	2.30	-6638.82	-0.02
13	MV U 1	I(13)	Moment-y	0.30	0.00	241.42	2.30	-7242.46	-0.02
14	MV U 1	I(14)	Moment-y	0.48	0.00	230.23	41.15	-7850.23	-0.19
15	MV U 1	I(15)	Moment-y	1.26	-0.01	753.13	-98.12	-8460.70	0.36
16	MV U 1	I(16)	Moment-y	0.59	-0.01	1025.96	-268.46	-11084.62	0.62
17	MV U 1	I(17)	Moment-y	-0.58	0.03	1906.69	146.43	-14005.03	-1.17
18	MV U 1	I(18)	Moment-y	1.01	0.03	-1063.61	509.56	-11707.35	1.80
19	MV U 1	I(19)	Moment-y	1.81	0.02	-914.48	241.77	-9753.36	1.87
20	MV U 1	I(20)	Moment-y	0.99	-0.03	-452.66	-392.62	-8316.63	-2.43
21	MV U 1	I(21)	Moment-y	0.41	-0.02	-294.20	-168.74	-7541.56	-1.60
22	MV U 1	I(22)	Moment-y	0.15	0.03	-208.67	-5.71	-7024.62	0.14
23	MV U 1	I(23)	Moment-y	0.04	0.03	-208.12	-1.80	-6504.59	0.08
24	MV U 1	I(24)	Moment-y	0.00	-0.01	-109.23	0.60	-6103.68	0.01
25	MV U 1	I(25)	Moment-y	0.00	-0.01	-109.21	0.57	-5830.90	0.04
26	MV U 1	I(26)	Moment-y	0.00	0.00	0	0	0	0
27	MV U 1	I(27)	Moment-y	0.00	0.00	0	0	0	0
28	MV U 1	I(28)	Moment-y	0.00	0.00	0	0	0	0
29	MV U 1	I(29)	Moment-y	0.00	0.00	0	0	0	0
30	MV U 1	I(30)	Moment-y	0.15	0.03	210.57	2.37	-5549.91	-0.45
31	MV U 1	I(31)	Moment-y	0.15	0.03	210.57	2.37	-5549.91	-0.45

Concurrent member forces

Figure 62. Moving Load Results

Checking Stresses due to Combined Loads

Create load combinations.

Results /  Combinations

PostCS

Name(**Temperature**) ; Type>**Envelop**

LoadCase> **Temperature (+)(ST)** ; Factor(**1.0**)

LoadCase> **Temperature (-)(ST)** ; Factor(**1.0**)

Name(**Top-Bot Temp Diff**) ; Type>**Envelop**

LoadCase> **Top-Bot Temp Diff (+)(ST)** ; Factor(**1.0**)

LoadCase> **Top-Bot Temp Diff (-)(ST)** ; Factor(**1.0**)

Name(**ULS 1**) ; Type>**Add**

LoadCase> **Summation(CS)** ; Factor(**1.15**)

LoadCase> **Erection Load(CS)** ; Factor(**1.2**)

LoadCase> **SM(SM)** ; Factor(**1.2**)

LoadCase> **MV U1** ; Factor(**1.0**)

Name(**SLS 2**) ; Type>**Add**

LoadCase> **Summation(CS)** ; Factor(**1.0**)

LoadCase> **Erection Load(CS)** ; Factor(**1.0**)

LoadCase> **Wind(ST)** ; Factor(**1.0**)

LoadCase> **SM(SM)** ; Factor(**1.0**)

LoadCase> **MV S 2 3** ; Factor(**1.0**)

Name(**SLS 3**) ; Type>**Add**

LoadCase> **Summation(CS)** ; Factor(**1.0**)

LoadCase> **Erection Load(CS)** ; Factor(**1.0**)

LoadCase> **Temperature(CB)** ; Factor(**1.0**)

LoadCase> **Top-Bot Temp Diff(CB)** ; Factor(**0.8**)

LoadCase> **SM(SM)** ; Factor(**1.0**)

LoadCase> **MV S 2 3(MV)** ; Factor(**1.0**)

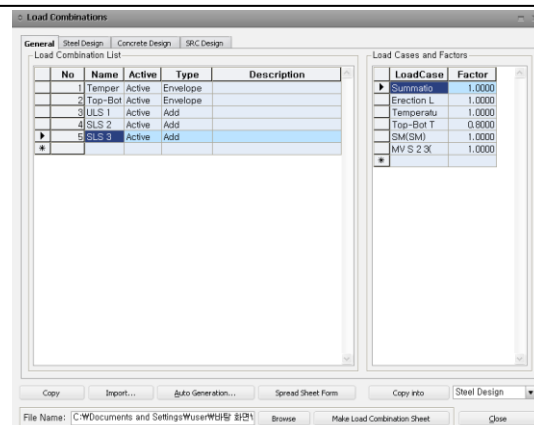



Figure 63. Creating Load Combinations

Check stress results due to load combinations.

Results / Stresses /  Beam Stresses(PSC)...
 Load Cases/Combinations>**CBall:SLS 3**
 Section Position>**Position 1**
 Components>**Sig-xx(Summation)**
 Type of Display
Contour (on) ; Legend (on)

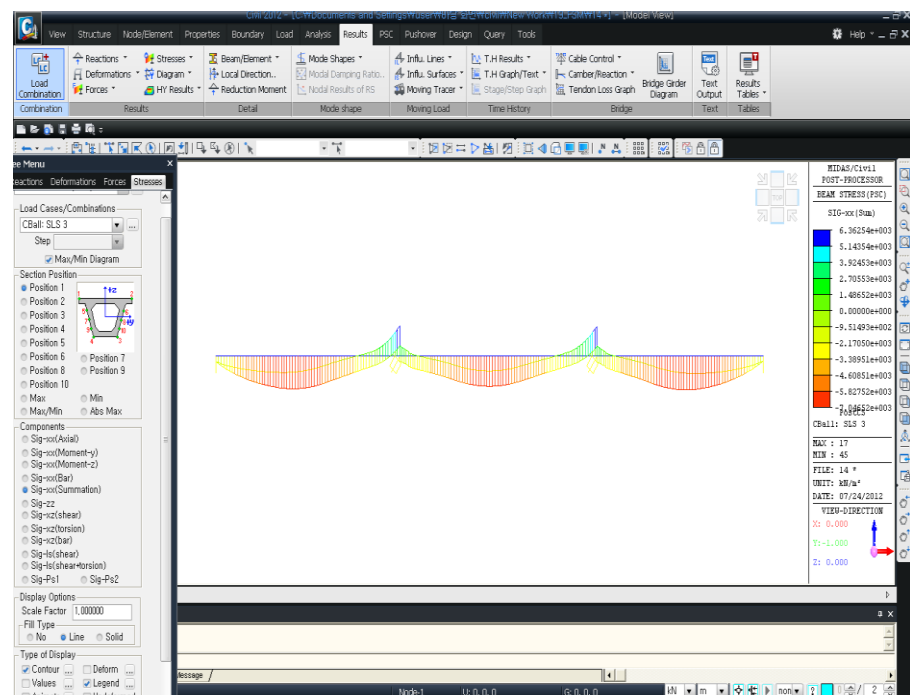


Figure 64. Stress Results due to Serviceability Limit State Combination 3